

40th Annual Armament Systems: Guns - Ammunition - Rockets - Missiles Conference & Exhibition "Translating Lessons Learned into Systems Requirements"

"Translating Lessons Learned into Systems Requirements"

25 -28 April 2005

Agenda

Tuesday, 26 April 2005

General Session:

- Weapons Systems and Explosives Safety in a Joint Warfighting Environment, Mr. David C. Schulte, Executive Director, Naval Ordance Safety & Security Activity
- The Future of Small Missiles, Dr. James C. Bradas, Associate Director for Missile Technology, AMERDEC

Session: Modeling & Simulation

- Accuracy Modeling of the 120mm M256 Gun as a Function Of Bore Centerline Profile, Mr. David Smith for Dr. Ronald Gast, Benet Laboratories
- Opening New "DOORS" to Managing JSF Gun System Requirements, Ms. Renee I. Bellack, General Dynamics Armament and Technical Products
- · Optimized Trajectory Shaping Guidance for an Air-to-Ground Missile Launched from a Gunship, Mr. Shane Sorenson, Naval Surface Warfare Center
- Estimating Ballistic Limits of Skin and Clothing for Projectiles, Mr. Henry E. Hudgins, US Army ARDEC
- A Review of The Insensitive Munitions Design Technology Workshop, Mr. M. Pascal Marchandin, NATO-MSIAC

Luncheon:

• Super Weapons - An Analysis, LTC Simon R. West, British Army, United Kingdom Defence Academy

Session: Medium Caliber System

- Oerlikon Ammunition for New Defense Environment, Mr. Allan N. Buckley, BTECH Oerlikon Contraves Pyrotec AG
- · Multi Mission Vehicle Armament & Air Burst Munition for Expeditionary Warfare Force Protection, Mr. Andrew Bradick, Lockheed Martin
- 25mm Gun Systems for the F-35 Joint Strike Fighter (JSF), Mr. Douglass C. Parker, General Dynamics Armament and Technical Products
- Optimized Gun Barrel Targeting Investigation, Mr. Jeff A. Siewart, Arrow Tech Association
- Mk 110 Mod 0 / 57mm Naval Gun & Ammunition Certification Process, LT Timothy J. Hackett, USCG, US Coast Guard Deepwater Sponsors' Representative
- GAU-19/A Barrel Life Study, Mr. James J. St. Germain, General Dynamics Armament & Technology Products
- RNLA IFV Firepower: 30 mm versus 35 mm 35 mm KETF Firing doctriney, Mr. Eelko van Meerten, TNO Defence, Safety & Security
- OMk44 Gun/Ammo IPT, Maj Kirk D. Mullins, USMC, DRPM AAA
- O30mm Airburst Development Translating Lessons Learned into System Requirements, Mr. Paul A. Reynolds, General Dynamics OTS

Wednesday, 27 April 2005

General Session:

- ATEC Update, BG(P) James R. Myles, USA, Commanding General, US Army Test & Evaluation Command
- Raytheon Missile Systems: A Global Perspective, Mr. Robert Salyer, Director, Business Development Raytheon Missile Systems
- · Acquisition and Sustainment Program, COL Lloyd E. McDaniels, USA, Project Manager, CCWS Project Office

Session: Weapon Systems

- Weapon System Concepts for a Future Gunship, Mr. Michael M. Canaday, Naval Surface Warfare Center
- · Royal Navy Small Calibre Gun Research to Defeat the Small Boat Threat, Mr. Johnathan Watkins, Defence Scientific Technology Laboratory
- Mini-Typhoon Remote Operated Small Arms Mount (ROSAM), Mr. Benjamin J. Hardie, General Dynamics Armament and Technical Products
- Update on Picatinny High Speed Turret, Mr. Mr. Richard Ciekurs, US Army RDECOM-ARDEC
- 40mm CTWS Supporting UK and France, Mr. Michael Duckworth, CTA International

Session: Missles & Rockets

Abraham Overview, Mr. Robert Daunfeldt, Bofors Defence

- Summary Overview of an Advanced 2.75 Hypervelocity Weapon, Mr. Larry Bradford, CAT Flight Services, Inc
- APKWS Flight Test Results, Mr. Larry Ingram and Mr. Dean Slocum, General Dynamics Armament & Technology Products
- APKWS Block II Demonstration Program, Mr. Milton E. (Gene) Henderson, Jr., US Army RDECOM-AMRDEC
- Missile Systems Lethality Enhancement Through the Use of a Conducting Aerosol Plasma Warhead, Mr. Allen H. Stults, US Army RDECOM-AMRDEC
- Next Generation Adaptable RF Seekers for Precision Munitions, Dr. Cory Myers, BAE Systems IEWS
- Technology for the Smart Rocket Launcher: The System Enabler For The 21st Century, Mr. Donald E. Davis, US Research, Development & Engineering Command
- Development of a Unique Penetrator Warhead for Rocket or Missile Delivery, Mr. Roger W. Melin, Lockheed Martin Missiles and Fire Control

Session: Large Caliber

- Development of the M1028, 120mm Anti-Personnel Tank Round, Mr. Hugh MacMillan, US Army Armaments Research, Development & Engineering Center and Mr. Neal Hylton, General Dynamics - Ordnance and Tactical Systems
- Metallic Materials & Processes Enabling Lightweight System Initiatives, Mr. Jeff Lehner, Director, Military Programs, Alcoa/Howmet Corporation
- Advanced Modular Gun Demonstrator: Redefining "Faster Than A Speeding Bullet", Mr. Steve Coladonato, Applied Ordnance Technology, Inc. The Modified Tank Ammunition IMI M152/6 HEAT-AP- T, Mr. Danny Schirding, Chief Systems Engineer
- Tank Ammunition Directorate IMI Ammunition Group
- A105/120/125 mm PELE Firing Results, Dr. Lutz Börngen, Rheinmetall Wafe Munition
- Line Of Sight/Beyond Line Of Sight (LOS/BLOS) Advanced Technology Demonstrator (ATD), Mr. David C. Smith, P.E., USA Benet Laboratories

Session: Energetics

- · Development, Evaluation and Lifetime Prediction of Medium and Large Caliber Ammunition, Mr. Gert Scholtes, TNO
- Concepts and Practices in Finding and Applying Lessons Learned, Mr. David F. Fair, US Army ARDEC
- Propellant Replacement for the 105-mm M67 Propelling Charge, Ms. Adriana L. Eng, US Army ARDEC
- Lead Azide Replacement Program, Mr. John M. Hirlinger, US Army RDECOM-ARDEC
- · Modeling Efforts for Autorotation Delivery System Concept Development, Mr. David C. Rutledge, Ph.D., Staff Engineer, United Defense

Thursday, 28 April 2005

General Session:

- · Direct Fire Ammunition Lessons Learned: "More Than Just Impacts on Bullets", COL Mark Rider, USA, Project Manager, Maneuver Ammunition Systems
- U.S. Army ARDEC Overview & Special Weapon Observation Reconnaissance Direct-Action System (SWORDS), Mr. Anthony Sebasto, Associate Director for Technology & Business Development, AETC
- National Defense Industrial Association (NDIA) Armament Division 2005 Division Status, Mr. Dave Broden, Chairman, Armaments Division, NDIA

Session: Technology & Manufacture

- Automated Ammunition Identification, Mr. David F. Pouliot, United Defense L. P.
- Design for Manufacturing & Assembly (DFMA), Mr. Steve Watts, US Army RD&E Command
- ARDEC Business Development Process, Mr. David L. Burkhardt, Director, Strategic Communications, US Army ARDEC
- MEMS IMU Common Guidance, Dr. Vicki C. LeFevre, US Army RDECOM-AMRDEC and Mr. David W. Panhorst US Army ARDEC, US Army ARDEC
- Development and Testing of High Explosive (HE) Projectiles for Electro-Magnetic Gun Army Tech Objective (ATO), Mr. Manfredi Luciano, US Army ARDEC
- Metal Injection Molding of Wing/Flaperon, Mr. Jerry C. LaSalle, Director of MIM Operations, Polymer Technologies, Inc (PTI)
- TBX Evaluation Testing in the M151 (2.75") Warhead as Risk Reduction for the APKWS, Mr. Jason C. Gilliam, US Army RDECOM-AMRDEC
- Archer Artillery Program, Mr. Ulf Einefors, Bofors Defence
- Improvements to Airborne Ladar Man-in-the-Loop Operations, Mrs. Sarah J. Hard, RDECOM-AMRDEC

Session: Mortars & Artillery

- M865 TID Improvement Study, Mr. Jason W. Gaines, General Dynamics
- Lessons Learned from the Development of the U.S. Navy 5-Inch Force Protection Projectiles, Mr. Sanford (Luke) Steelman, III, Naval Surface Warfare Center
- Advanced Gun Barrel Technologies, Dr. Amir Chaboki and Mr. Allen Boutz, United Defense
- Defining Homogeneity for Medium Caliber Ammunition and Small Grain Propellant Lots, Mr. Scott Carney, ATK
- Precision Fires for the Field Artillery, Mr. John Halvey, Raytheon, and Stefan Blomgren, Bofors Defence
- Low Cost Course Correction (LCCC) Demonstration Program, Mr. George Barnych and Mr. Daniel Davis, Ordance and Tactical Systems Division
- XM395 Precision Guided Mortar Munition (120mm PGMM): Responsive, Standoff Precision Lethality for Highly Deployable and Mobile Forces, Mr. James Terhune and Mr. Anthony Pezzano, OPM Mortars
- Precision Guided Miniature Munitions, Mr. Mark Carlson, BAE Systems
- The 81mm Non Lethal Mortar Carrier Projectile (MoCaP), Mr. Seungeuk Han, Mr. Andrew Ponikowski, and Mr. Raymond Trohanowsky, US Army RDECOM-ARDEC
- Commercial Disposal of Explosive Wastes, Mr. Mark M. Zaugg, EBV Explosives Environmental Company



"Translating Lessons Learned into Systems Requirements"

40th Annual Armament Systems: Guns - Ammunition - Rockets - Missiles Conference & Exhibition

April 25 - 28, 2005

Sheraton New Orleans Hotel New Orleans, LA



Monday, April 25 2005

10:00 a.m. On-site Registration

Noon Exhibit Move-In

5:00 p.m. - Reception in the Exhibit Hall

6:30 p.m.

6:30 p.m. Adjourn for the Day

Tuesday, April 26, 2005

7:00 a.m. On-site Registration / Continental Breakfast

7:45 a.m. Opening Remarks

8:00 a.m. *Mr. David C. Schulte*, Executive Director, Naval Ordnance Safety & Security Activity

8:30 a.m. *Dr. James C. Bradas*, Associate Director for Missile Technology, AMRDEC

9:00 a.m. **Session: Modeling & Simulation**

Accuracy Modeling of the 120mm M256 Gun as a Function of Bore Centerline Profile

Mr. David Smith for Dr. Ronald G. Gast, Benet Laboratories

Opening New DOORS to Managing JSF Gun System Requirements

Ms. Renee I. Bellack, General Dynamics Armament and Technical Products

Optimized Trajectory Shaping Guidance for an Air-to-Ground Missile Launched from a Gunship

Mr. Shane Sorenson, Naval Surface Warfare Center

9:30 a.m. Exhibit Hall Opens

10:00 a.m. Break in the Exhibit Hall

10:30 a.m. Conceptual Weapon System Design for the Defense of Naval Vessels from the Swarming Small Boat Threat

Mr. John E. Bibel, Naval Surface Warfare Center Dahlgren Division

Warhead Penetration Dynamics - Warhead Body, Fuze, and Target Interaction

Mr. Richard Ventura, Talley Defense Systems

Ballistic Limits of Skin and Clothing for Lethality Estimates of Projectiles Wound Ballistics

Mr. Henry E. Hudgins, US Army ARDEC

A Review of the Recent NIMIC IM Design Technology Workshop

Mr. M. Pascal Marchandin, NATO - MSIAC

11:50 a.m. Luncheon: Super Weapons From a Historical and Psycological Basis

LTC Simon R. West, British Army, United Kingdom Defence Academy

1:10 p.m. **Session: Medium Caliber Systems**

Ammunition for the New Infantry Battelfield Environment

Mr. Allan N. Buckley, BTECH Oerlikon Contraves Pyrotec AG

Force Protection - Multi Mission Vehicle Armament & Air Burst Munition for Expeditionary Warfare *Mr. Andrew Bradick*, Lockheed Martin

F-35 Joint Strike Fighter Gun Overview and System Update

Mr. David L. Maher and Mr. Douglass C. Parker, General Dynamics, Armament and Technical Products

Phalanx Targeting Investigation

Mr. Jeff A. Siewart, Arrow Tech Association

3:00 p.m. Break in the Exhibit Hall

3:30 p.m. Session: Medium Caliber Systems (Continued)

Mk 110 Mod 0 / 57mm Gun Test & Certification Process

LT Timothy J. Hackett, USCG, US Coast Guard

GAU-19/A Barrel Life Study

Mr. James J. St. Germain, General Dynamics Armament & Technology Products

Calibre Choice for the Dutch IFV

Mr. Eelko van Meerten, TNO Defence, Safety & Security

The Expeditionary Fighting Vehicle, How Operational and Combat Lessons Learned Apply to the EFV and the 30MM Mix of Tomorrows Warfighter

Maj Kirk D. Mullins, USMC, DRPM AAA

30x173mm HEAB-T Development and Lessons Learned

Mr. Paul A. Reynolds, General Dynamics - OTS

5:30 p.m. -

Reception in the Exhibit Hall

7:00 p.m.

7:00 p.m.

Adjourn for the Day

Wednesday, April 27, 2005

7:00 a.m. On-site Registration / Continental Breakfast

7:45 a.m. Opening Remarks

8:00 a.m. **BG (P) James R. Myles, USA**, Commanding General, US Army Test & Evaluation Command

8:45 a.m. *Mr. Robert Salyer*, Raytheon

9:10 a.m. Acquiring and Sustaining US Army Missiles

COL Lloyd E. McDaniels, USA, CCWS Project Office

9:30 a.m. Exhibit Hall Opens

9:45 a.m. Session: Medium Caliber Systems (Continued)

Recent Developments of the M230 30MM Chain Gun *Mr. Lawrence A. Mason*, ATK Ordnance & Ground Systems

MK44 Automatic Cannon Update

Mr. Mark McMillian, ATK Ordnance Systems

10:30 a.m. Break in the Exhibit Hall

Concurrent Sessions

10:50 a.m. **Session: Weapon Systems**

AC-130U Gun System Production Re-Start *Mr. John G. Fletcher*, General Dynamics Armament and Technical Products

Weapon System Concepts for a Future Gunship

Mr. Michael M. Canaday, Naval Surface Warfare Center

Royal Navy Small Calibre Gun Research to Defeat the Small Boat Threat *Mr. Johnathan Watkins*, Defence Scientific Technology Laboratory

11:50 a.m. - Lunch

1:00 p.m. Session: Weapon Systems (Continued)

Remote Operated Small Arms Mount (ROSAM)

Mr. Benjamin J. Hardie, General Dynamics Armament and Technical Products

Placing Gunner's Behind the Proctective Armor of Vehicles

LTC Kevin P. Stoddard, USA, PM Soldier Weapons

Picatinny High Speed Turret (PHIST)

Mr. Richard Ciekurs, US Army RDECOMARDEC

Session: Missiles & Rockets

Critical Asset Defense - ABRAHAM Rocket Assisted Projectile

Mr. Robert Daunfeldt, Bofors Defence

Hypervelocity Propulsion System Substantially Improves 2.75 Rocket Lethality, Safety, Survivability *Mr. Larry Bradford*, CAT Flight Services, Inc.

APKWS Flight Test Results

Mr. Larry S. Ingram, General Dynamics Armament and Technical Products

APKWS Block II Demonstration Program *Mr. Milton E. (Gene) Henderson, Jr.,* US Army RDECOM-AMRDEC

Lunch

Session: Missiles & Rockets (Continued)

Missile System Lethality Enhancement Through the Use of Pulsed Power and Plasma Conduction *Mr. Allen H. Stults*, US Army RDECOM

Next Generation Adaptable RF Seekers for Precision Munitions

Dr. Cory Myers, BAE Systems

The Smart Rocket Launcher as the Key Enabler for the Rocket System of the Future: The Technology Developments Needed for the Next Generation Rocket Laucher to Carry 70mm Rockets into the 21st Century *Mr. Donald E. Davis*, US Army Research, Development & Engineering Command

Session: Weapon Systems (Continued) Session: Missiles & Rockets (Continued) Recent Activities Involving 40mm CTWS in Development of a Unique Penetrator Warhead for Rocket Support of UK and France or Missile Delivery Mr. Michael Duckworth, CTA International Mr. Roger W. Melin, Lockheed Martin Missiles and Fire Control The Marine Corps Expeditionary Fire Support Determining the Army Aviation Rocket and Missile Mix for System (EFSS): A Systems Overview the Future Fight Mr. Jason Burkett, General Dynamics Mr. William M. Mulholland, Whitney, Bradley & Brown 2:40 p.m. Break in the Exhibit Hall Break in the Exhibit Hall (Last Opportunity to Visit Exhibits) (Last Opportunity to Visit Exhibits) 3:00 p.m. Exhibit Hall Closed Exhibit Hall Closed 3:10 p.m. Session: Large Caliber **Session: Energetics** Development of the XM1028, 120mm Anti-Advanced Propelling Solutions Complying with Demands Personnel Tank Round (FCS) Mr. Hugh MacMillan, Armaments Dr. Beat Vogelsanger, NITROCHEMIE Wimmis AG Engineering and Technology Center, Mr. Peter *Georgantzis*, US Army ARDEC, and *Mr*. *Neal Hylton*, General Dynamics-OTS Titanium Investment Casting Weapon System Development, Evaluation and Lifetime Prediction of Medium Application and Large Caliber Ammunition Mr. Jeff Lehner, Director, Military Programs, Mr. Gert Scholtes, TNO Alcoa/Howmet Corporation Advanced Modular Gun Demonstrator - XLT Concepts and Practice in the Application of Lessons Test Gun Learned *Mr. Steve Coladonato*, Applied Ordnance Mr. David F. Fair, US Army ARDEC Technology, Inc. The Modified Ammunition, Equipped with the Propellant Replacement for the 105-mm Artillery Propelling "Fuzaman": The IMI 105-mm Heat-AP-t Charge Cartridge M152/6 Ms. Adriana L. Eng., US Army ARDEC Mr. Danny Schirding, Israel Military Industries, Ltd 105/120/125 mm PELE Firing Results Environmentally Benign Substitute Compounds for Lead Dr. Lutz Borngen, Rheinmetall Waffe Azide Munition Mr. John M. Hirlinger, US Army RDECOM-ARDEC Lightweight Gun Development and Testing for Modeling Efforts in Support of PKERS Concept the Future Combat System Development Mr. David C. Smith, P.E., USA Benet Mr. David C. Rutledge, Ph.D., Staff Engineer, United Laboratories Defense 5:15 p.m. Adjourn for the Day Adjourn for the Day

Thursday, April 28, 2005

7:00 a.m. On-site Registration / Continental Breakfast

7:45 a.m. Opening Remarks

8:00 a.m. *COL Mark Rider, USA*, Project Manager, Maneuver Ammunition Systems

8:30 a.m. *Mr. Anthony Sebasto*, Associate Director for Technology & Business Development, AETC

9:00 a.m. *Mr. Dave Broden*, NDIA Armaments Division Status Overview, Chairman, Armaments Division, NDIA

9:20 a.m. Break

Concurrent Sessions

9:40 a.m. **Session: Technology & Manufacture**

Automated Ammunition Identification *Mr. David F. Pouliot*. United Defense L. P.

Deep Digger Weapons System Concept *Mr. David W. Burns*, US Army ARDEC

Development of Composite Launch Tubes for Shoulder Fired Weapons through Applied Science, Planning and Teamwork *Mr. Thomas P. Jacobson*, Talley Defense Systems

Design for Manufacture & Assembly (DFMA) *Mr. Steve Watts*, US Army RD&E Command

Technology and Manufacturing Initiatives *Mr. Dave Burkhardt*, Enterprise Management Office, US Army ARDEC

Session: Technology & Manufacture (Continued)

Army MEMS Common Guidance Program Mr. David. W. Panhorst, US Army ARDEC and Dr. Vicki C. LeFevre, AMRDEC

Development and Testing of HE Projectiles for EM Gun - STO Mr. Manfredi Luciano, US Army ARDEC Session: Mortars & Artillery

Tank 120mm Training Ammunition: MB65 Target Impact Dispersion Study *Mr. Jason W. Gaines*, General Dynamics-OTS

Lessons Learned from the Development of the U.S. Navy 5-inch Ship Self Defense Projectiles *Mr. Sanford L. Steelman, III*, Naval Surface Warfare Center

ONR's Advanced Gun Barrel Technology Program *Mr. Allen Boutz*, United Defense

Structural Margin Improvement on the M829A3 Projectile

Mr. Scott Carney, ATK

Session: Mortars & Artillery

Excalibur: Turning the Field Artillery into a Long Range Precision Attack Weapon System *Mr. Stefan Blomgren*, Bofors Defence

Low Cost Course Correction (LCCC)
Demonstration Program *Mr. George B. Barnych*, General Dynamics

Ordnance and Tactical Systems Division

Session: Technology & Manufacture (Continued)

Advanced Metal Injection Molding Technology Applications to Defense Industry

Mr. Jerry LaSalle, Polymar Technologies

11:40 a.m. Lunch

1:00 p.m. **Session: Technology & Manufacture** (Continued)

TBX Evaluation Testing in the M151 (2.75") Warhead as Risk Reduction for the APKWS *Mr. Jason C. Gilliam*, US Army RDECOMAMRDEC

Advanced Precision Kill Weapon System *Mr. Ulf Einefors*, Bofors Defence

Test Results of an Imaging LADAR Seeker for Small Missiles *Mrs. Sarah J. Hard*, RDECOM-AMRDEC

Session: Mortars & Artillery (Continued)

Applying Six Sigma Principles to Implementation of the PGMM Training Concept Mr. Anthony Pezzano, OPM Mortars

Lunch

Session: Mortars & Artillery (Continued)

Precision Guided Miniature Munitions *Mr. Mark Carlson*, BAE Systems

Development of a Non-Lethal Mortar Delivery System

Mr. Seungeuk Han, US Army RDECOM-ARDEC and *Mr. Andrew Ponikowski*, US Army, RDECOM-ARDEC

Explosive Waste Recycle and Disposal *Mr. Mark M. Zaugg*, EBV Explosives Environmental Company

3:00 p.m. **Conference Adjourns**



Development of a Unique Penetrator Warhead for Rocket or Missile Delivery

Presented to:

National Defense Industrial Association 40th Annual Armament Systems: Guns - Ammunition - Rockets – Missiles (GARM)

Conference & Exhibition

25 – 28 April 2005

Roger W. Melin

Lockheed Martin Missiles and Fire Control roger.melin@lmco.com (972) 603-1769

- I-NAIL™ Penetrator Concept
- I-NAIL™ Penetrator Design
- Recent Testing
 - Penetration Tests
 - Static Expulsion Tests
 - Wind Tunnel Expulsion Tests

I-NAIL™ Project Introduction

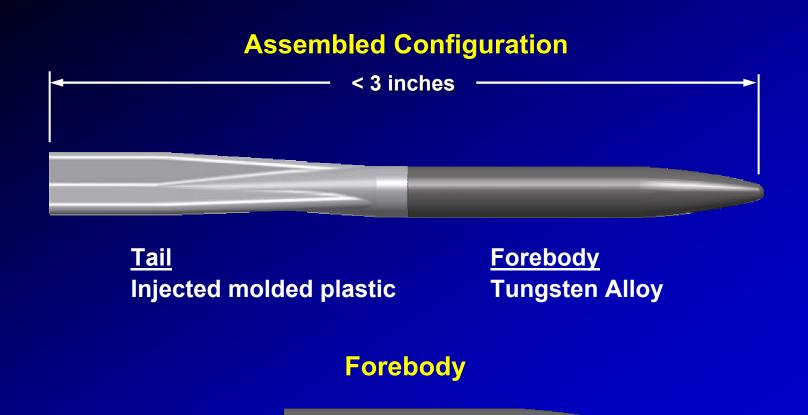


Project began as alternate GMLRS/HIMARS payload

- ✓ Zero dud rate
- ✓ Inexpensive
- ✓ Increased lethality
- ✓ Limited zone of effects
- Alternate platforms & applications
 - ✓ Hydra-70
 - **✓ APKWS**
 - ✓ AC –130 Gunship (105 mm cannon)







I-NAIL™ Penetrator Design Trades



Missiles and Fire Control

Testing

Forebody Materials

- Ceracom 2
- Ceracom 3 hipped
- Ceracom C116
- Ceracom not hipped
- French Sintered Rod
- French Swaged Bar
- Hawk (Formulas 1 3)
- HD17 Tungsten Bar
- HD17D Tungsten Bar
- Liquid Metal
- Sintered Tungsten
- Tungsten Welding Rod

Tail Materials

- Aluminum
- Magnesium
- Plastic
- Mischmetal (cerium & lanthanum)

Penetrator Masses

• 150 – 300 grains

Target Materials

- AI 5083
- AI 6061T6
- A36 Steel
- High Hard Armor
- Cast Iron Engine Manifold
- Concrete block
- Cinderblock Wall Simulant
- Flak Jacket
- Ballistic Gelatin

Impact Velocities

• 750 - 2000 f/s

Analysis

Penetrator Masses

• 150 – 300 grains

Forebody Geometry

- Nose Shape
 - Circular Ogive
 - Von Karman (3:1 - 1:1)
- Shaft Cross Section
 - Circular
 - Hexagonal
- Tip Radii
 - Flat
 - Hemispherical

Impact Velocities

• 750 - 2000 f/s



Business Development / Demo Tests

- Performed in conjunction with tungsten evaluations
- Variety of targets, penetrator designs, and impact conditions

Engineering Tests

- Performed to develop structured database
- Design of Experiments techniques used to design test matrix
- Results used to develop regression-based penetration predictors

LMMFC Light Gas Gun Facility

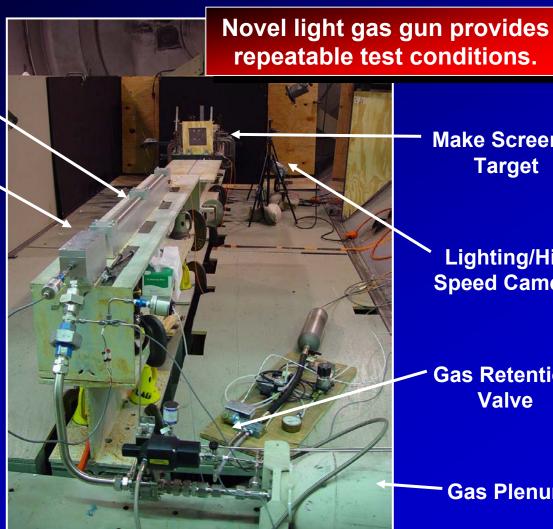


Gas (He) Gun

Breech



Data Acquisition/Data Reduction System



Make Screens/ **Target**

Lighting/High **Speed Cameras**

Gas Retention Valve

Gas Plenum

Representative Targets















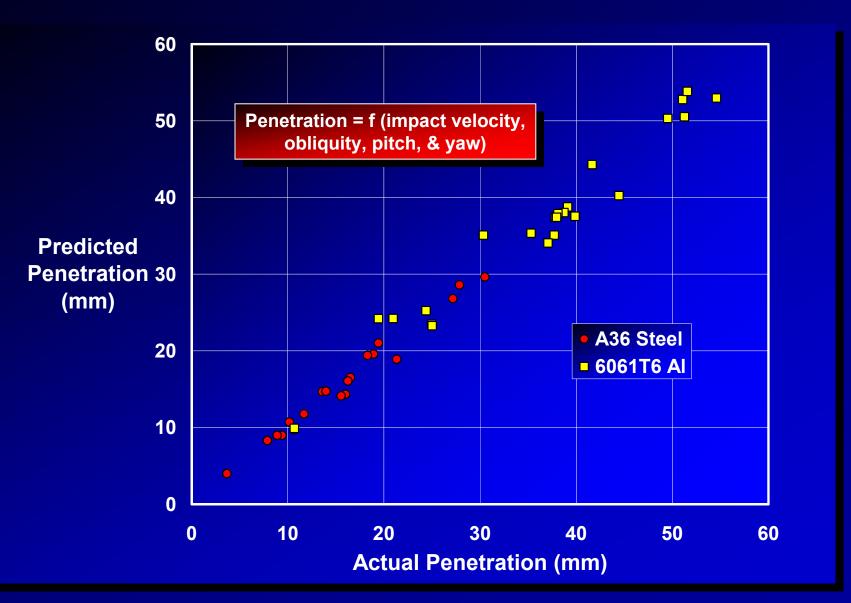
I-NAIL™ Sabot Separation



Make Screen Holder

I-NAIL™ Penetration Modeling





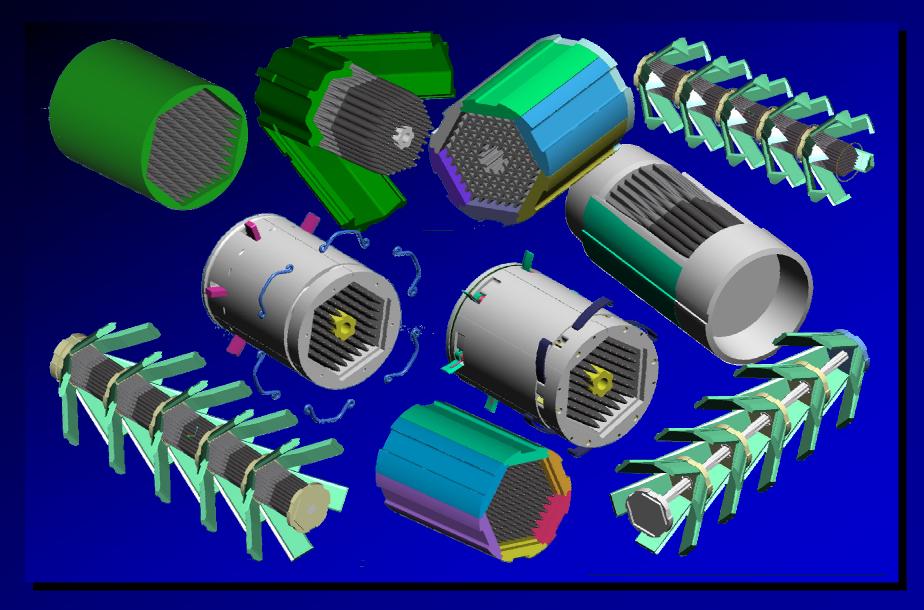


Objectives:

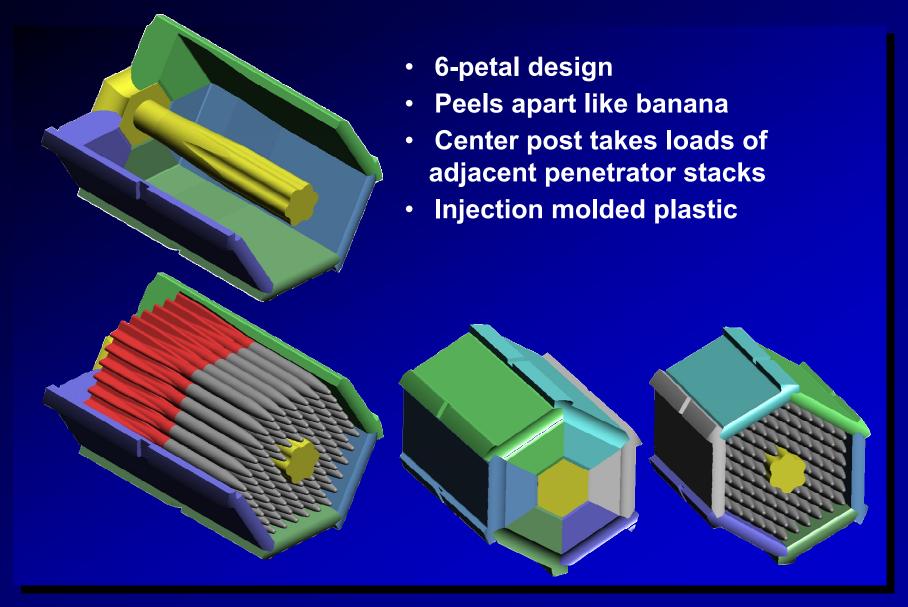
- Package maximum loadout of I-NAIL[™] penetrators maintaining HYDRA-70 weight / CG requirements
- Design and demonstrate performance of dunnage / penetrator support mechanism
- Demonstrate successful expulsion of I-NAIL™ penetrator payload with Hydra-70 expulsion charge
- Expulsion velocity ~150 f/s

I-NAIL™ Penetrator Dunnage Concepts Missiles and Fire Control





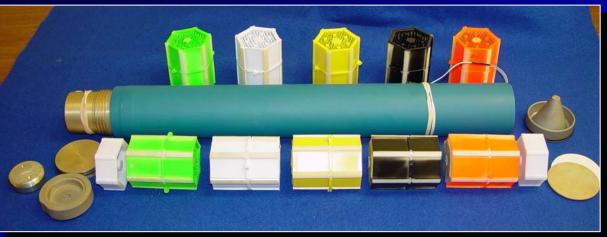
Selected Hydra-70 Dunnage Concept Missiles and Fire Control



Expulsion Test Hardware









- 390 I-NAIL[™] penetrators/warhead + 30 simulants for mass matching
- Fore & aft spacers added for CG match
- 6-Petal dunnage design for support and penetrator release
- GFE Hydra-70 expulsion charge
- Special SAF to allow static function

- Two Hydra-70 warhead casings loaded at Camden, AR facility with I-NAIL™ penetrators
- Two warhead tests performed on 20 October 2004 at National Technical Systems site in Camden, AR
- Static fired two warheads
 - No representative rocket airflow
 - No spin
- Three high-speed digital cameras used for data acquisition (2.1K frames/sec)
- Celotex package positioned down range for possible pattern data

Expulsion Test Layout





Down-Bore Views

















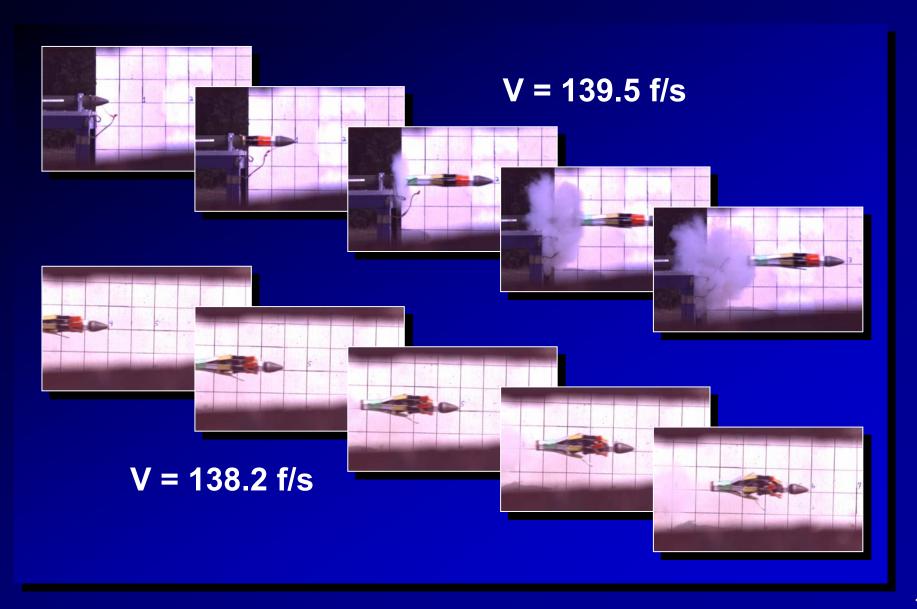






Side View - Test 2





Expulsion Test Results Summary

Missiles and Fire Control

- Both payloads successfully ejected
- Nominal ejection velocities achieved in both tests

Test 1: Camera 1 – no data

Camera 2 – 138.3 f/s

Test 2: Camera 1 – 139.5 f/s

Camera 2 - 138.2 f/s

- Most penetrator damage occurred from sideways impacts as opposed to expulsion event
- Actual flight conditions will minimize such effect since penetrators will have time to align correctly
- Penetrator ballistics as expected



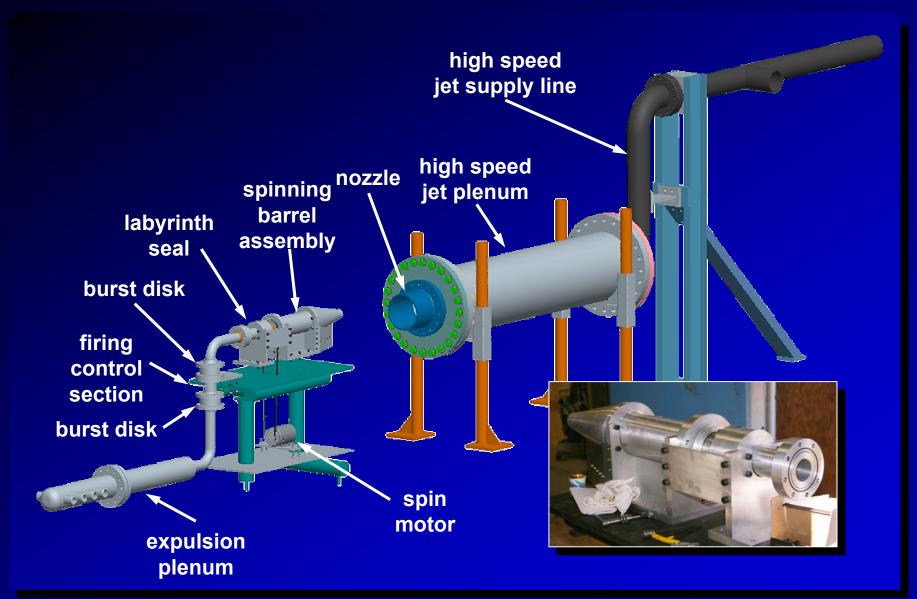
Objectives:

- Demonstrate separation cleanliness of two potential I-NAIL[™] dunnage designs
 - 6-Petal Design (Hydra-70)
 - 3-Compartment Design (APKWS)
- Gather initial conditions for possible use in future dispense and pattern simulation studies

- Testing performed at LMMFC High Speed Wind Tunnel (HSWT) facility in Grand Prairie, TX on 18 December 2004
- "Backyard" Tests High velocity flow ducted out of high pressure tanks to external test location
- Spinning air gun constructed to expel payload into high mass flow air stream
- Payloads represented two I-NAIL[™] penetrator pack concepts
 - 5 packs present in M255-A1 Hydra-70
 - 3 packs present in APKWS

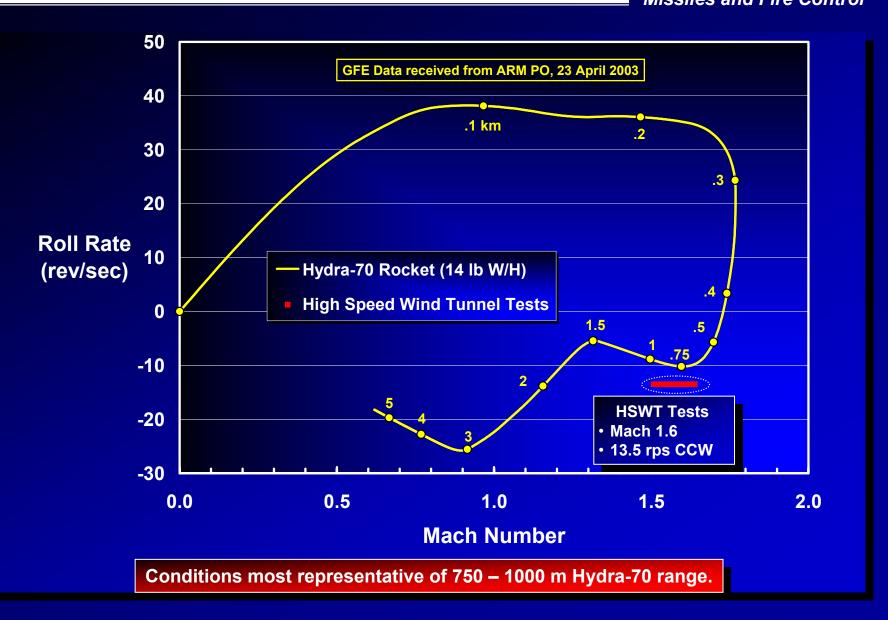
Wind Tunnel Test Setup





I-NAIL™ Wind Tunnel Test Conditions

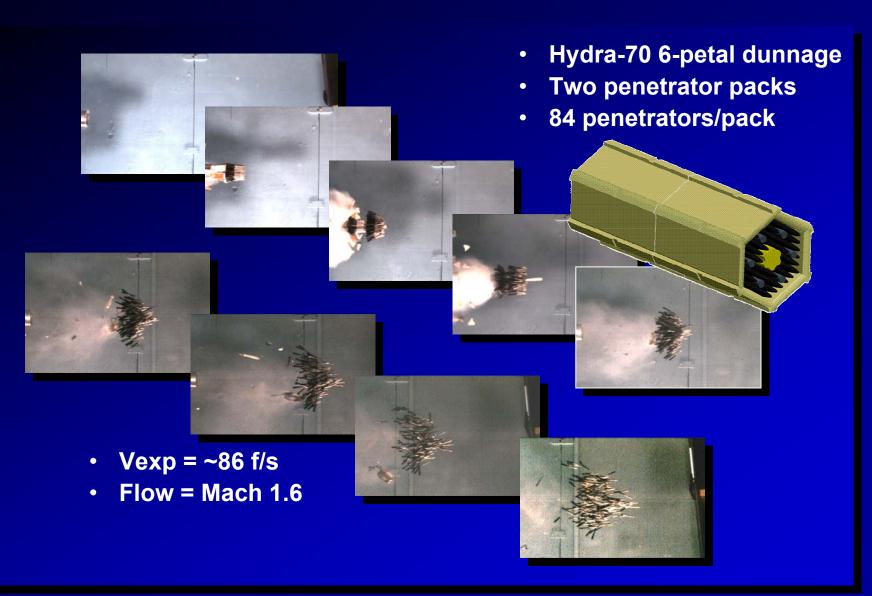




I-NAIL™ Wind Tunnel Test 1



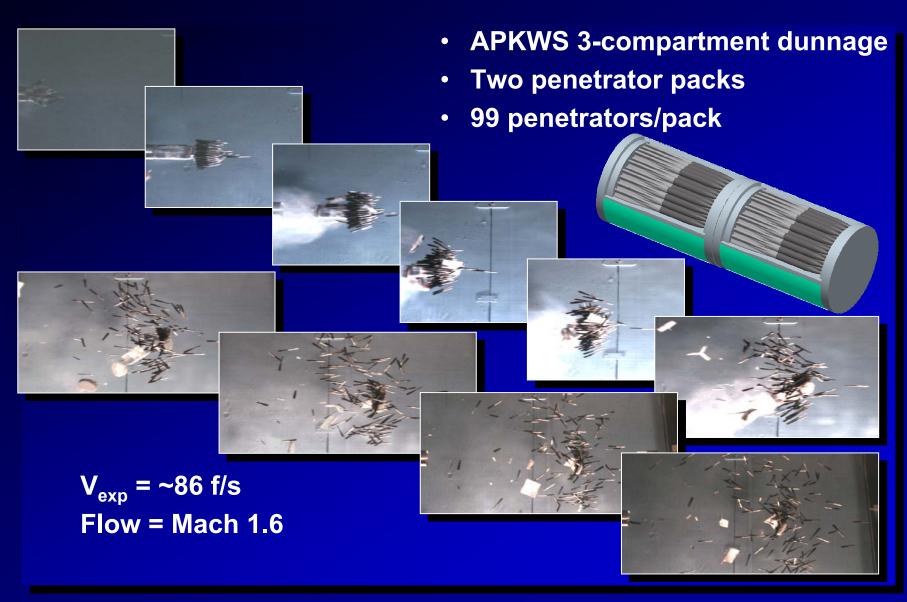
Missiles and Fire Control



I-NAIL™ Wind Tunnel Test 2



Missiles and Fire Control





6-Petal Dunnage Concept

- Good release achieved
- Petals broke in a desired fashion and moved away quickly

3-Compartment Dunnage Concept

- Center structure interferes with radial dispense of penetrators
- Compartment covers and solid forward plate are pushed into penetrator cloud

Both Concepts

- Collisions occurred between two penetrator packs
- Second pack catches up to first mainly due to still being pushed by plenum gas; drafting effects may contribute
- Good penetrator dispersion and aerodynamics

Wind Tunnel Test Conclusions



<u>Dunnage</u>

- 6-Petal dunnage design preferred
 - Demonstrated better overall performance
 - Compatible with Hydra-70 and APKWS platforms
 - Utilizes existing M255-A1 components
 - Inexpensive solution for APKWS

Penetrators

 Design has been modified to strengthen weak point in tail attachment section to minimize breakage

Viable dunnage concept has been tested and is ready for integration and flight testing.

- Mini-penetrator design developed
- Design provides significant behind-armor effects
- Highly lethal with no unexploded ordnance left on the battlefield
- System integration approach and implementation demonstrated
- Compatible with a variety of delivery systems





Next Generation
Adaptable RF
Seekers for
Precision
Munitions

40th Annual Guns-Ammunition-Rockets-Missiles Conference

Missiles & Rockets Session

April 27, 2005

Dr. Cory Myers
BAE Systems IEWS
cory.s.myers@baesystems.com

Mission Need





- Provide small unit of operations with organic Precision Strike capability against High Value Targets
- Accelerate Enemy Defeat
- Reduce Collateral Damage
- Improve Deployability & Logistics
- RF Guided Munition (RFGM)



Completes the sensor-to-shooter chain for IO targets operating from 30MHz to 3GHz



Current Mortar Munitions generally do not achieve first shot direct hit on target. RFGM guidance system capable of correcting trajectory improves first-shot hit on the target to 50%.

BAE SYSTEMS

System Concept

- Exploit dismounted, close-in attack scenario with small aperture, RF seeking weapon
 - If the dismount (SOF) can be cued to the presence of the emitter then the dismount can attack the (soft target) emitter with an organic weapon (e.g. 81 mm mortar)
- Create a passive, all-weather, and inexpensive precision RF seeker capability for multiple weapon types
 - Enable a suite of precision and area suppression weapons (ground-to-ground, ground-to-air, and air-to-ground) that home on RF energy all using similar RF seeker and guidance technology
- Deny enemy use of RF spectrum for military purposes
 - Counter enemy radar/IR/acoustic signals Camouflage,
 Concealment and Deception (CCD) efforts

DARPA Hard Technical Challenge: Quick and Precise Geo-location of RF Emitters from a Single, High-Velocity, Small Weapon

BAE SYSTEMS

Technical Challenges

System Requirements:

- Quick: Geo-location estimate must be fast enough (5 sec) to guide a mortar which has only 25-30 seconds of flight time
- Precise: Geo-location with an objective radius of an 81 mm mortar (20 m)
- RF Emitters: Target frequencies from 30 MHz to 3 GHz and multiple waveforms
- Single: Emissions received by only a single platform (passive technique)
- High-Velocity: Velocity of a mortar varies from 300 m/sec to 100 m/sec
- Small: e.g. 81 mm mortar form factor restricts antenna size and distance

Technology Enablers:

- Organic detection (cueing) capability
- Small, lightweight, wideband, and inexpensive RF receivers
- Inexpensive memory and processors
- Proliferation of guided weapons (IR, laser, GPS, etc.)

DARPA RFGM Program

BAE SYSTEMS

- Replacement fuze/guidance package that effectively converts current, ballistic 81 mm mortar munitions into precision RF guided munitions
- Screw-on mod-kit
- Affordable, Easy to use
- Frequency range 30MHz to 3GHz
- Accuracy not dependent on visual observation
- Fire and Forget
- Passive, all-weather
- Technology that is scalable to other munitions

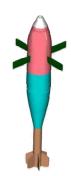






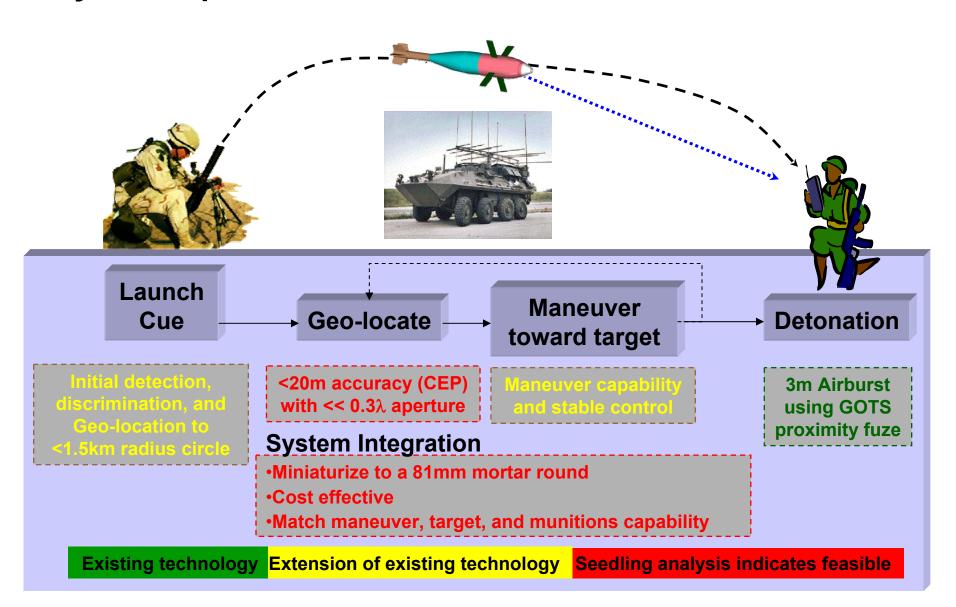






BAE SYSTEMS

System Operation

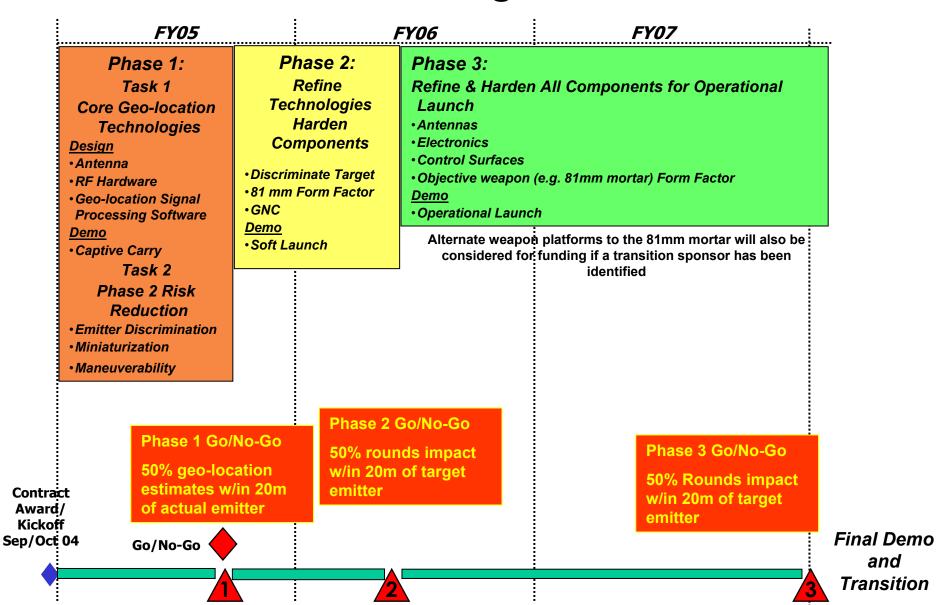


Design/Trade Space

- Cueing:
 - The weapon receives cueing information from an external system such as Wolfpack, ACS, etc.
 - Utilize SIGINT standard emitter descriptors (carrier frequency, bandwidth, modulation, etc.) to future proof weapon versus template matching emitter waveforms
- Geo-location
 - Despite high SNR condition, classic DF techniques alone will not work well enough due to the limited aperture size/spacing and the (low) frequency range of interest
- Maneuver toward target
 - Guidance/control techniques are well known (e.g. ERGM, PGMM, etc.)
- Detonation
 - Utilize existing GOTS fuze technology to avoid re-qualification costs
- System Integration
 - Optimizing the relationship between geo-location accuracy and aerodynamic control authority while minimizing weight, volume, and cost and impact on weapon range and effects
 - · Integrating the RF Guided Munition kit with the fuze is preferred
 - Volume/length will need to be added to the weapon (mortar) for antennas, RF electronics, signal processing, and control surfaces in a manner that minimizes range loss
 - Using GPS is possible but an IMU may be sufficiently capable while being cheaper than SASSM modules – both add a precise targeting capability

RF Guided Munitions Program



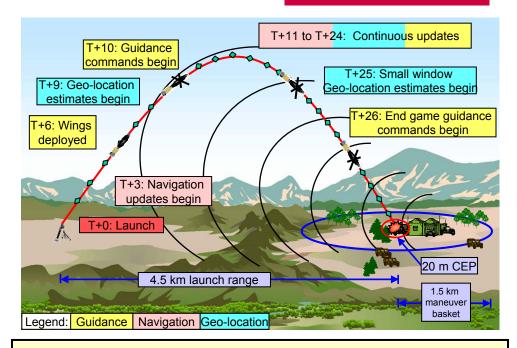


Geo-location Challenge

BAE SYSTEMS

Geo-location Error Sources:

- Thermal noise
- Quantization noise
- Phase noise
- Receiver spurs, intermods and harmonics
- Man-made noise and atmospheric noise at HF
- Navigation errors from position and roll sensors
- Channel mismatch errors
- Calibration errors
- Multi-path signal corruption
- Co-channel signal interference
- Platform motion induced modulation

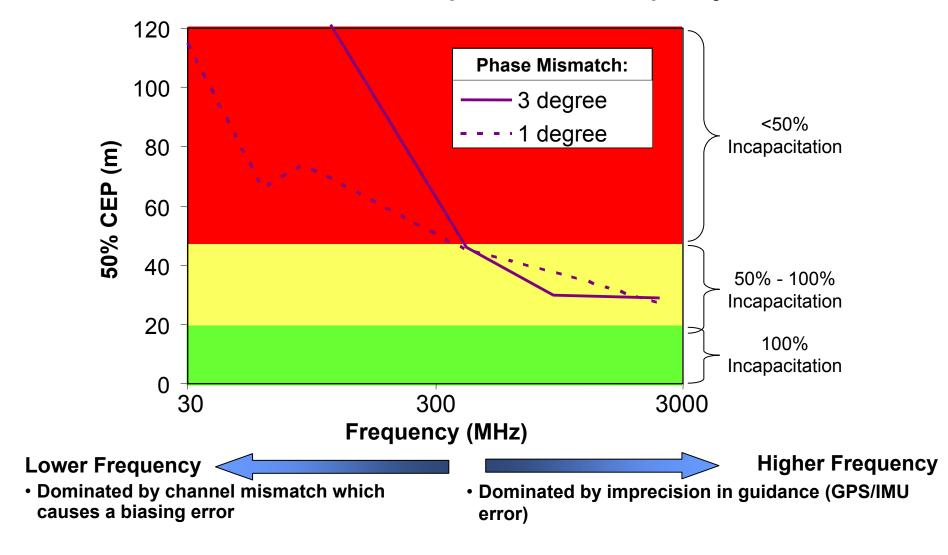


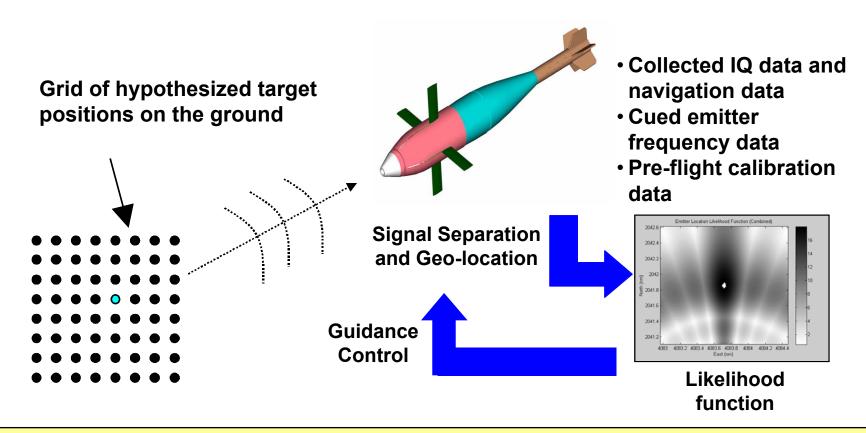
Geo-location Requirements:

- Provide guidance commands well before apogee to support maneuver basket.
- Deal with multi-emitter environment.
 Guide to one emitter, not the centroid of emitters.
- Provide resiliency to multi-path and polarization.

Geo-location Challenge

Angular precision of classic DF techniques is limited by λ /D, SNR, and channel mismatch which is unacceptable for low frequency emitters

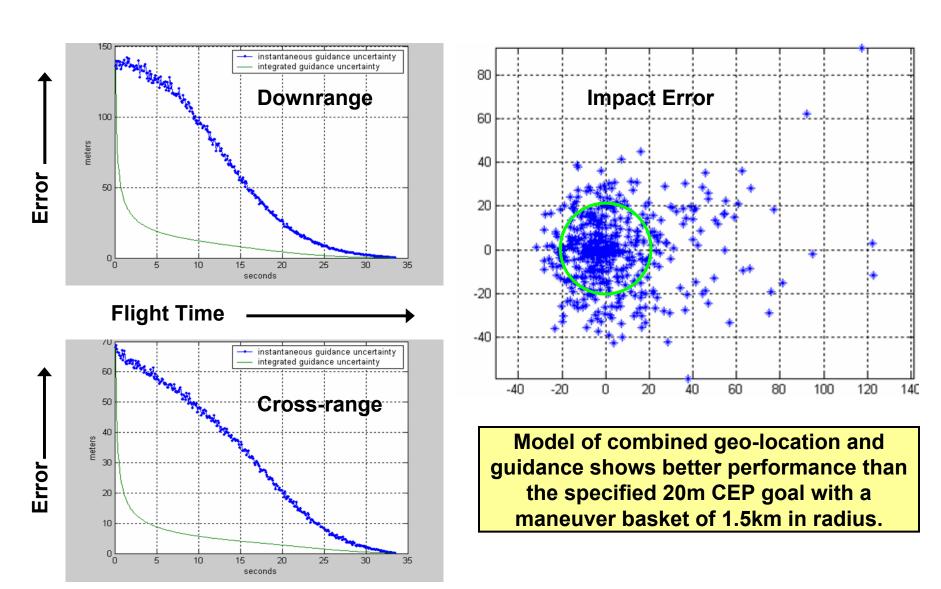




Geo-location method uses temporal, phase and amplitude information from all the antenna elements, separates signals of interest and then determines emitter geo-location metric by computing the probability likelihood surface of the potential emitter location as a function of its hypothesized location.



Geo-location and Guidance Performance

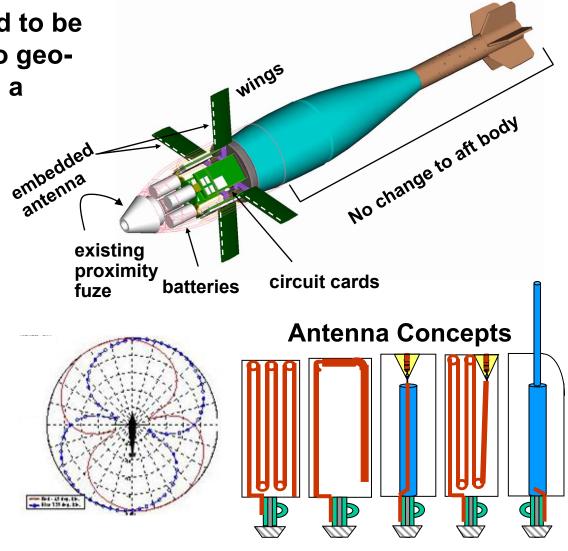


System Integration

BAE SYSTEMS

Multiple subsystems need to be integrated, in addition to geolocation, to make RFGM a reality:

- -Antennas
- -Receivers
- -Actuators
- -Wings
- Navigation
- -Guidance
- -Control
- -Signal Processing
- -Power
- -Cueing
- -Fuze





Questions?

Points of Contact

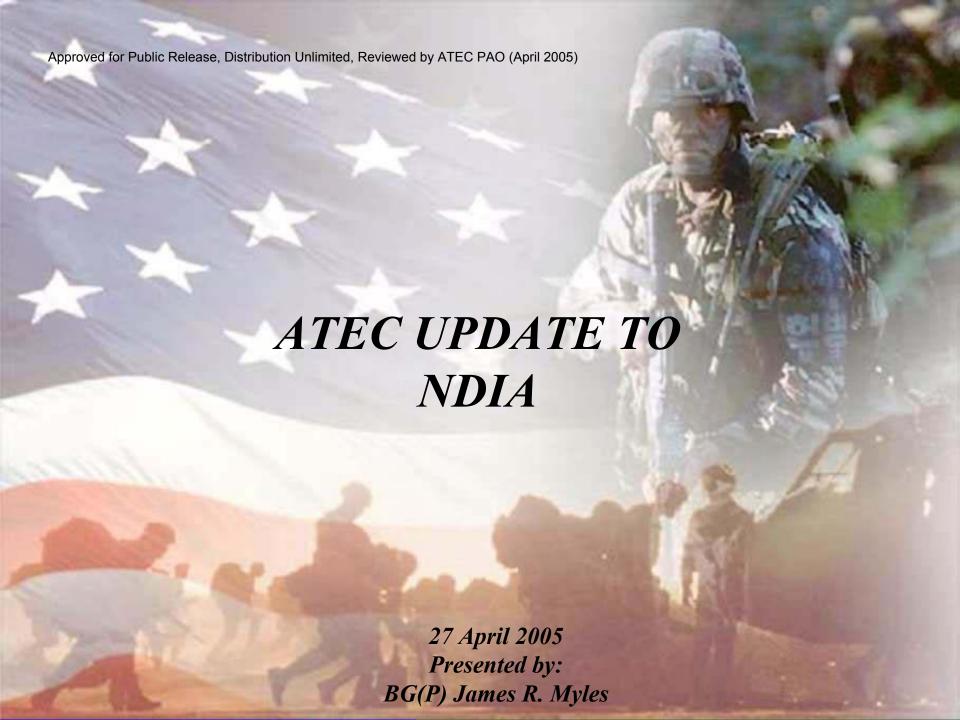
BAE SYSTEMS

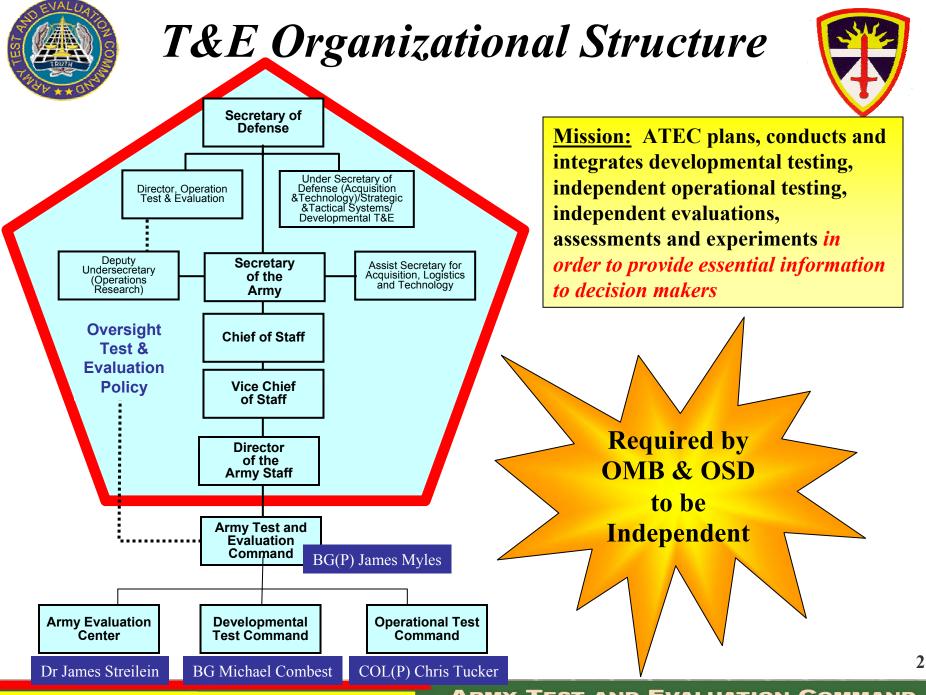
DARPA/ATO Program Manager Dr. John Allen jallen@darpa.mil

BAE Systems Program Manager Ms. Marianne Tenore marianne.tenore@baesystems.com Phone: 603-885-8470

BAE Systems Management Dr. Cory Myers cory.s.myers@baesystems.com Phone: 603-885-6845

BAE Systems Business
Development
Mr. Daniel Bradford
daniel.bradford@baesystems.com
Phone: 603-885-5937





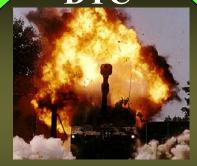




The Army's T&E is Unique



DTC



Developmental Testing engineering type tests conducted under controlled conditions

ATEC

AEC



Evaluation
independent assessment of all
testing and simulation

OTC



Operational Testing testing conducted using real soldiers in simulated combat

ATEC integrates developmental and operational testing

4



ATEC T&E Philosophy



- Testing is part of acquisition process
- Two Fundamental Missions
 - Make system better (test-fix-test)
 - Provide Info to Decision Makers



- Does it Work...How do I know?
- Evaluation in Depth:

 Platform → System of Systems → Unit Mission
- No Pass/Fail → Capabilities and Limitations
 Good Enough ≠ Anything will do











- OTs with fewer troops tougher and shorter
- Rapid Acquisition: Spiral Development, DT/OTs,...
- System of systems T&E
- Support to War is the norm
- OTs During Major Training Events (NTC, JRTC,...)





GWOT -- What is ATEC Doing About It



- War Support
 - Early Involvement with PMs & REF
 - Focus on performance in the AOR
 - Soldiers in DT / Limited User Testing
 - Respond to Warfighter!
 - -- Up-Armor Vehicles, Slat Armor, Stryker, ECMD, Robotics,...
 - Deploy ATEC Assessment Team in Theater
- Safety Confirmations
- Capabilities and Limitations Report
 - For the Commander
 - What do we know (Cap and Limits)
 - What don't we know
 - Safety, C/L, Training, Supportability and Accountability





Back-up Slides



Additional Thoughts



- NOT YOUR DAD'S ATEC
 - LIKE THE ARMY, ATEC STANCE AND BALANCE HAS CHANGED
 - FOCUS IS GWOT AND SOLDIERS IN HARMS WAY TODAY
 - TIMELINES HAVE SHRUNK TO SUPPORT RAPID ACQUISITION...STANDARDS HAVE NOT!
 - DATA VOIDS ARE FILLED WITH OUR EXPERIENCE AND BEST MILITARY JUDGEMENT
- WE TEST-FIX-TEST AND PROVIDE INFO TO OUR SENIOR LEADERS TO MAKE DECISIONS
- WE ARE NOT SEPARATE FROM THE ACQUISITION COMMUNITY
- GET US INVOLVED EARLY
- WE DO DT/OT
 - UNDERSTAND THE SUBTLETIES...WILL LOOK AT SOMETHING BEFORE IT IS READY...WE KNOW THE DIFFERENCE.
 - BECAUSE WE SEE THAT IT WORKS IN DT DOESN'T MEAN IT WORKS WITH SOLD!
- LOGISITICS, TRAINING, ACCOUNTABILITY OF THE SYSTEMS ARE SHOWSTOPPERS FOR OUR SOLDIERS.
 - LACK OF TRAINING AND LOGISTICS PROVIDES BAD REPUTATION FOR THE SYSTEM...HARD TO REMOVE THE STIGMA
- NDAA 03
- PUT THE SOLDIER FIRST AND YOU ARE NEVER WRONG
- MOMS AND DADS OF AMERICA HAVE EXPECTATIONS
 - THAT THEIR SON AND DAUGHTER RETURN HOME TO THEIR FAMILY
 - THAT THEIR SOLDIER HAS THE BEST EQUIPMENT ON THE FACE OF THIS EARTH
 - AND, THAT WHEN NEEDED, IT WILL WORK...THAT IT WILL WORK



Modeling Efforts for Autorotation Delivery System Concept Development

Presented at the 40th Annual Guns & Ammunition/ Missiles & Rockets Symposium & Exhibition

April 25-28, 2005 New Orleans, LA

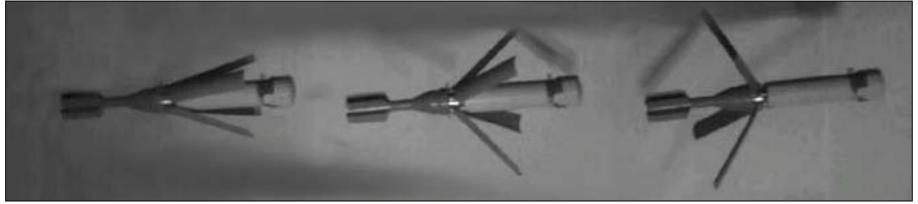
David C. Rutledge, Ph.D., Staff Engineer, United Defense, L.P. Mark Costello, Ph.D., Oregon State University



^{*} Autorotation Delivery Systems is patent pending

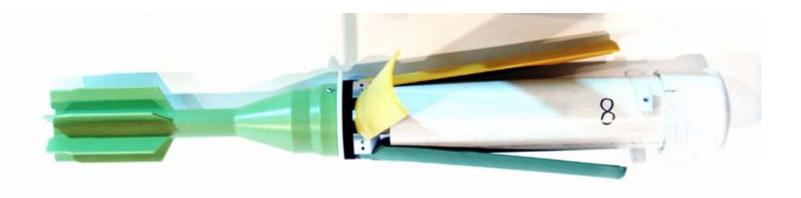
- Overview
- Deployment Sequence
- Applications
- Modeling Performed
- Axisymmetric Model
- BOOM Model
- Summary and Video





Overview

- The Autorotation Delivery System, formerly known as Projectile Kinetic Energy Reduction System (PKERS), is a concept developed by United Defense as an autorotation decelerator for high-value tactical payloads
- Combines a projectile body with a deployable rotor that reduces descent velocity via autorotation
- Modeling and simulation will facilitate the optimal design process during each stage of development



Deployment Sequence

- Rotors stowed conformal to the sides of projectile body prior to deployment
- During deployment, rotors rapidly rotate outward due to projectile spin and aerodynamic drag
- Transition to autorotation occurs as the rotor blades become aerodynamically loaded coupled with an increasing spin rate
- System attains a steady descent velocity when the inertial and aerodynamic forces reach equilibrium



Applications

- United Defense is developing the Autorotation Delivery System as an alternative to conventional parachutes for certain applications
- Flight characteristics and descent velocities are tailorable for different missions and payloads (e.g., land and sea sensors, cargo, battle damage assessment, munitions)
- Can be gun launched, mortar launched, or air dropped
- Modular design allows accommodation of all the necessary components required for precision guidance



First Generation Autorotation
Delivery System integrated with
Talley SMAW-D Motor

United Defense

Modeling Performed

- An axisymmetric spreadsheet-based model was developed to estimate the dynamics and loads as the rotors initially open and impact the damper
- A detailed flight mechanics model was developed and integrated into the BOOM smart weapon simulation system to model the flight dynamics from initial rotor deployment to full autorotation

United Defense

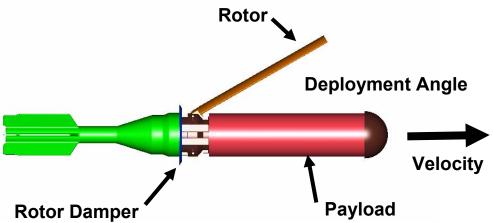
Axisymmetric Model:

Initial Deployment

- The Axisymmetric Model is a quasistatic spreadsheet-based concept development tool
 - Allows estimates of system performance and parametric studies
 - Assumes symmetry about the longitudinal spin axis, zero spin rate, and conservation of rotor angular momentum
 - Provides 2 Degrees of Freedom (DOF) for the projectile body (forward velocity and spin) and 1 DOF for each rotor blade (deployment angle)

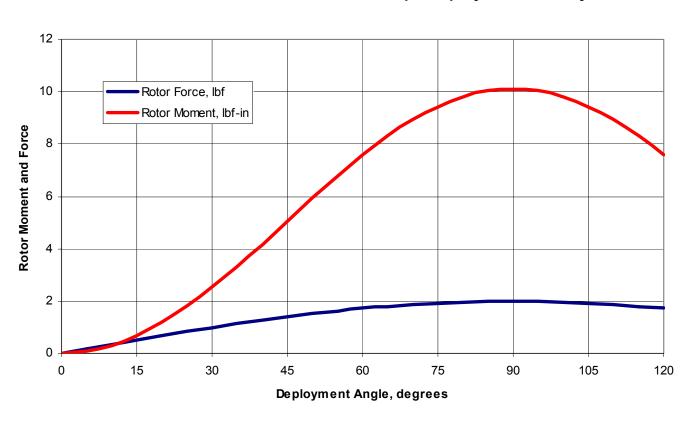
Aerodynamic Load Calculation

- Axisymmetric model assumptions:
 - Centrifugal loads due to flight element spin are a minor contributor during initial rotor opening (zero spin assumed)
 - Flight element velocity constant during initial deployment (worst case)
 - Aerodynamic drag is then a function of deployment angle
- Calculate upper limit on rotor force and moment about rotor hinge as a function of deployment angle



Aerodynamic Loads on Rotor

Rotor Force and Moment at 100 fps Deployment Velocity



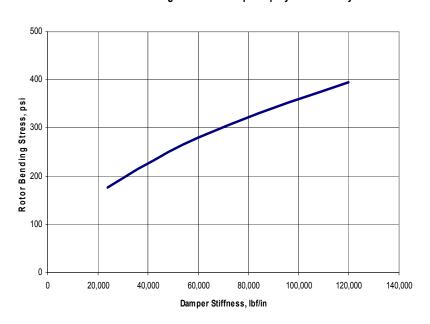
Calculation of Damper Force

- Worst case rotor loads accelerate the rotor open until they impact the damper at approximately 120 degrees
- Opening moment is numerically integrated versus deployment angle to get the angular momentum at initial impact with damper
- Corresponding kinetic energy is absorbed by the damper
- Maximum damper force is calculated for multiple damper locations, shapes, and materials to optimize design

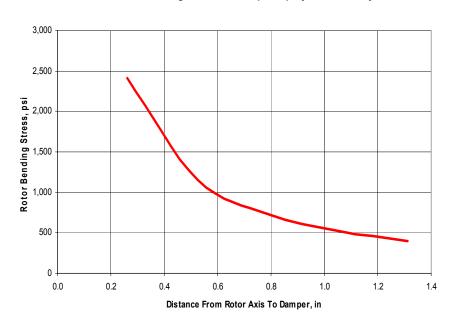
Rotor Bending Stress Results







Rotor Bending Stress at 100 fps Deployment Velocity



Worst case force is used to design rotors so they can safely survive the bending stress during initial deployment

Rotor Deployment Video



Camera 2: 0 – 12'



Camera 3: 12 – 24'



Camera 4:

24 - 36'



United Defense

BOOM Model:

Detailed Flight Dynamics

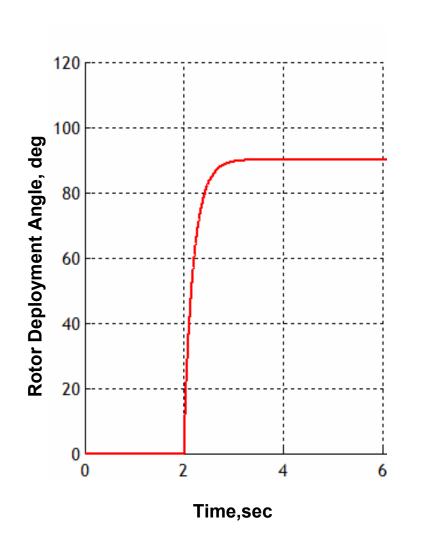
- The BOOM model is a detailed design and development tool
 - BOOM is a smart weapon simulation system
 - Full model of the delivery system flight mechanics was developed and integrated into BOOM
 - Provides 6 rigid-body DOF for the projectile body (flight element center of gravity position and orientation angles), 1 DOF for each rotor blade (deployment angle), and a 3-state rotor dynamic inflow representation

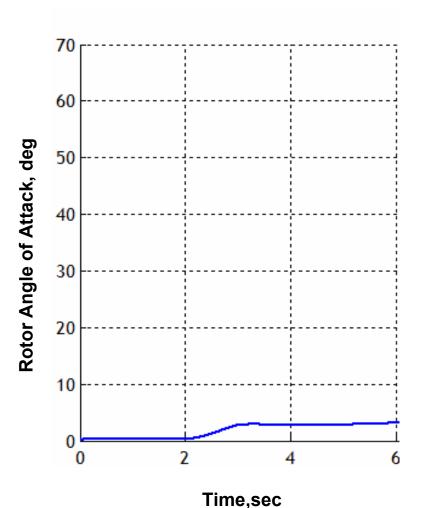
United Defense

BOOM Model Description

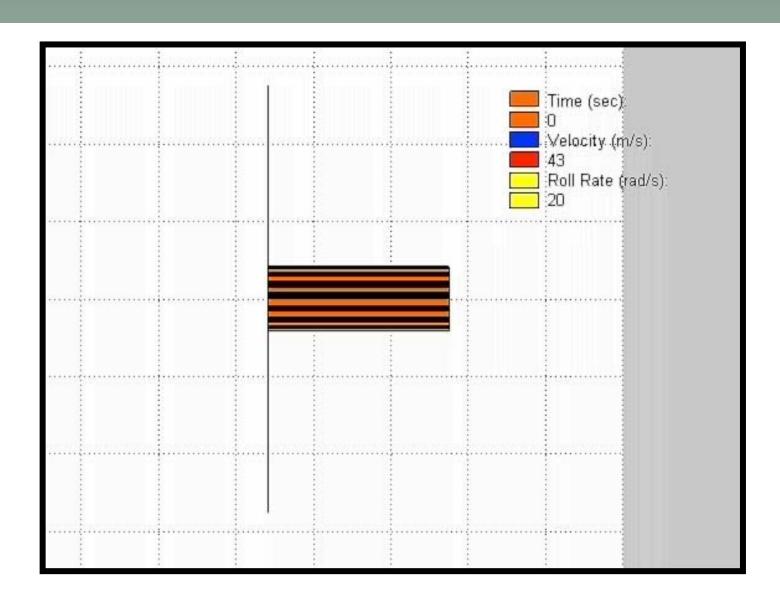
- Aerodynamic loads on the projectile body are a function of angle of attack and Mach number
- Aerodynamic forces and moments on the rotors are computed using blade element theory; airfoil lift and drag coefficients are a function of local rotor section angle of attack and Mach number
- BOOM simulations then provide the dynamics of the system as a function of time
- The simulation presented here has the following initial conditions:
 - Linear velocity 43 m/s (141ft/s)
 - Spin (Roll) rate 20 rad/s (3.2 rev/s)
 - Deployment angle fixed at 90 degrees

BOOM Model Rotor Kinematics





BOOM Model Top View Animation



United Defense

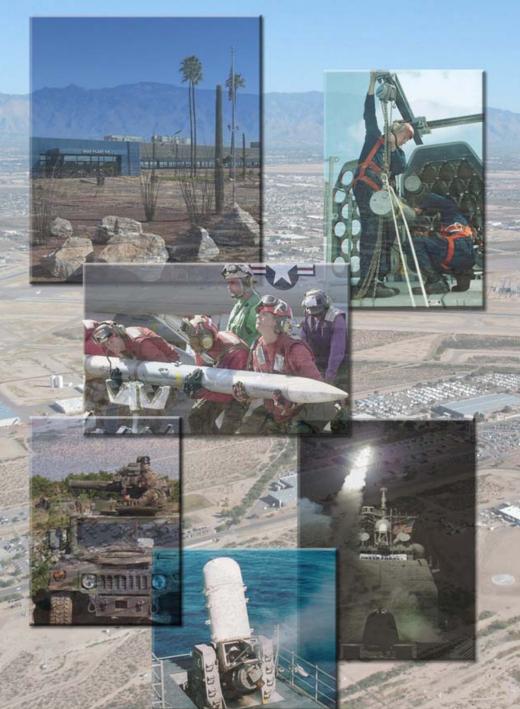
BOOM Model Results

- BOOM model results indicate that the concept will deploy in a manner consistent with flight tests
- Model has not been validated yet. Validation with test data is planned for the future, making the model capable of supporting future design, flight control system development, and payload integration
- This simulation assumes that all rotors have the same deployment angle at any time; this causes a short numerical instability that's not present in the actual system
- Model to be modified to allow each rotor to have different deployment angles

Summary and Video

- Test configurations have proven to be robust enough to survive deployment stresses
- Additional development, testing, and demonstration is planned to validate the 6-DOF model and applications
- Exploring a variety of Payloads and Applications
 - Sensors, Cameras, Munitions
 - Reconnaissance, Surveillance, Repeaters





Raytheon

Customer Success Is Our Mission

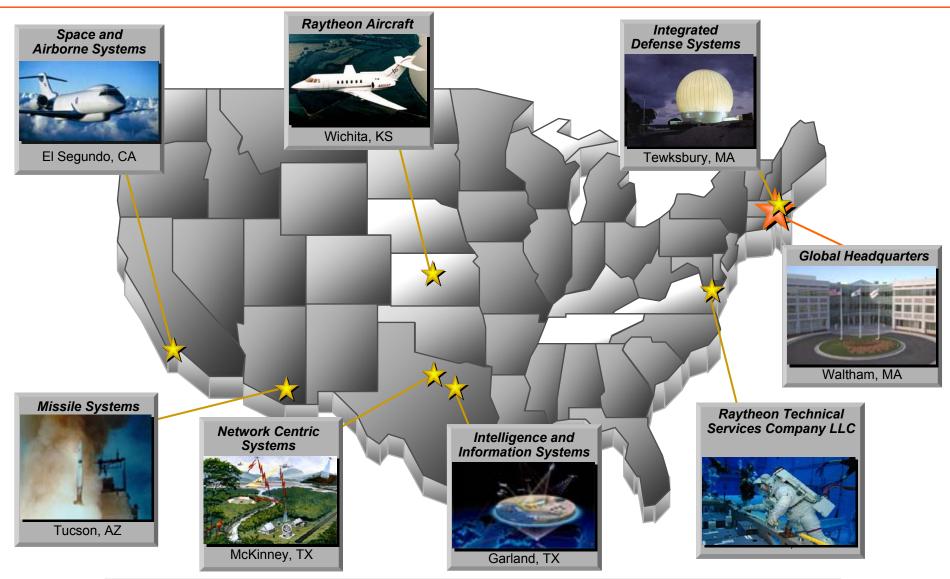
Raytheon Missile Systems: A Global Perspective

Robert D. Salyer
Director, Business Development
Raytheon Missile Systems

NDIA Symposium April 27, 2005



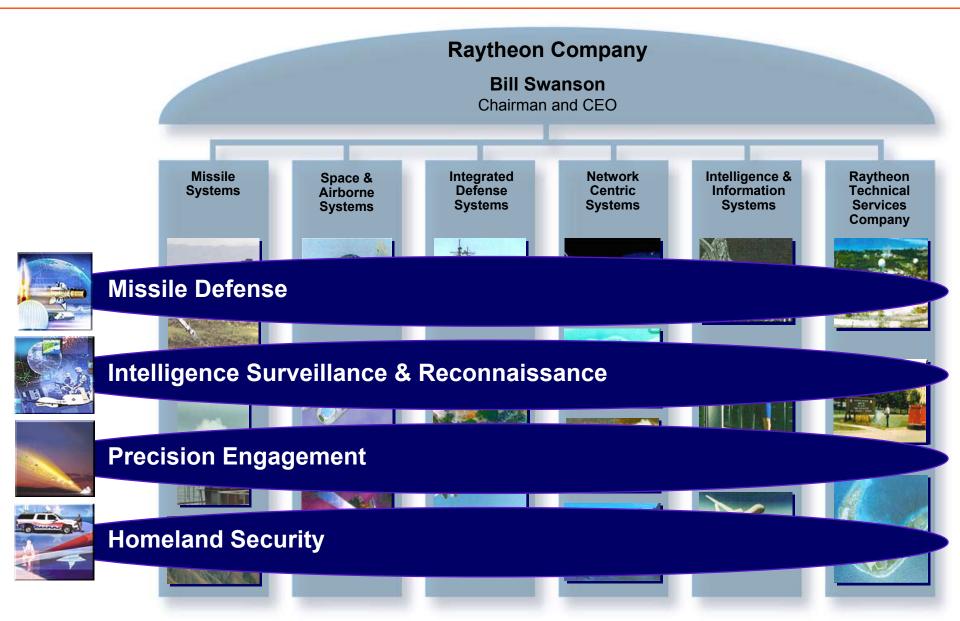
Raytheon Company



80,000 Employees; 2004 Revenue: \$20.2B



Business / SBA Intersection



Customer-Focused Marketing



- Meet our commitments
- Actively seek every opportunity to proactively work with our customers to define their needs
- Develop and provide the best solutions
- Earn the customer's confidence

Customer Must View Us As a Valued "Partner of Choice"



Raytheon Missile Systems – Who We Are

- 2004 sales: \$3.8 billion
- 11,000 employees
- Headquartered in Tucson, Arizona
- World's largest developer, producer and integrator of weapon systems
 - More than 1 million missiles produced since 1954
 - 70% domestic; 30% international
- Broad weapons portfolio
 - Missiles
 - Smart munitions
 - Projectiles
 - Kinetic intercept vehicle
 - Directed energy weapons
- Customers: all U.S. military services;
 Allied Forces of more than 40
 countries



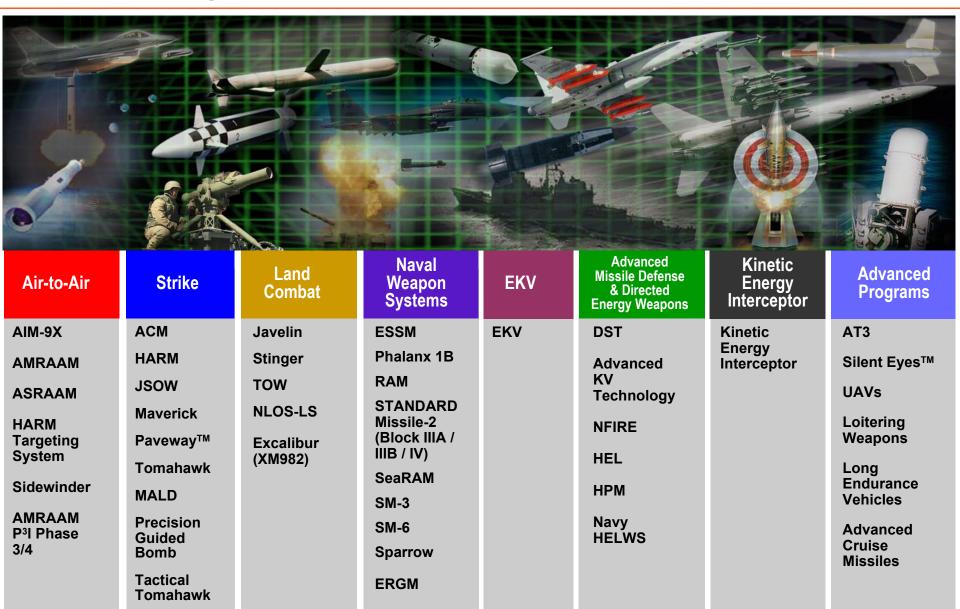


Our Vision





Missile Systems



50345A-7



Comparative Defense Budgets -- 2005

• US: \$401B

Germany: \$31B

• UK: \$53B

Australia: \$13B

• Japan: \$46B

South Korea: \$20B

Note: All Budget Figures above in \$US



How Defense Sells Into International Market

- Foreign Military Sales
- Direct Commercial Sales
- International Traffic in Arms Regulations
- Congressional notification



International Challenges

- Buy European/Buy America
- Lack of integration into U.S. markets
- Technology transfer
- Offsets desire for "noble" work
- Fluctuating exchange rates





Enablers

- Desire for U.S. products/technology
- Workshare opportunities
- Innovative contract structures
- Co-development opportunities
- Economies of scale reduce cost of U.S. production



Win-win Solutions Attractive to Buyers



Industry Response

- Grow international presence
 - Raytheon International Inc.
 - Regional in-country expertise
 - Business development/program teams on the road
 - Visibility at international trade shows/events
- Joint ventures
 - Diehl Raytheon Missile Systeme
 - Thales Raytheon
- Joint development opportunities
 - ESSM
 - Excalibur
 - RAM
- Co-production agreements





Looking into the Future

- Future "netted" battlespace
 - "Missile as a Node in the Net"
- Expanding into new markets
 - Directed energy
 - NASA space exploration
 - Guided Projectiles
 - Total life cycle logistics support
- Requires system engineers/ system architects



Expanding the Core Beyond the Missile Market



RMS Guided Projectile Family

Excalibur



Mission

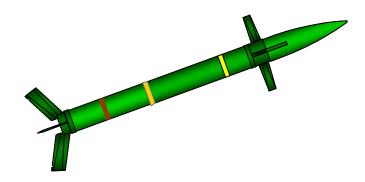
Indirect fires for legacy, interim and objective force

Paladin, XM777 and NLOS Cannon

➤ Extended range munition → 39 Cal > 37 Km → 52 Cal > 47 Km

➤ Precision guided, <20m CEP

Extended Range Guided Munition



Mission

➤ Naval Surface Fire Support

➤ DDG81 MK45 MOD4 (5") Gun

➤ Cruiser Conversion

➤ Extended Range Munition

➤ + >41 Nmi

➤ Precision Guided, <20m CEP



Phalanx Overview

Primary Mission:

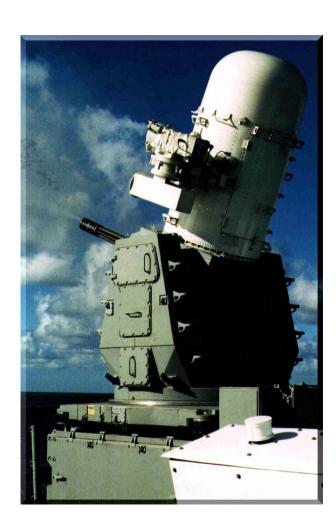
Terminal Defense Against ASCMS and High Speed Aircraft Penetrating Other Fleet Defensive Envelopes

Added Missions:

- Surface Mode
 - Counter Small, Fast Surface Craft and Slow Flying Helicopters and Aircraft
- Sensor Support For Close-in Missile Engagements

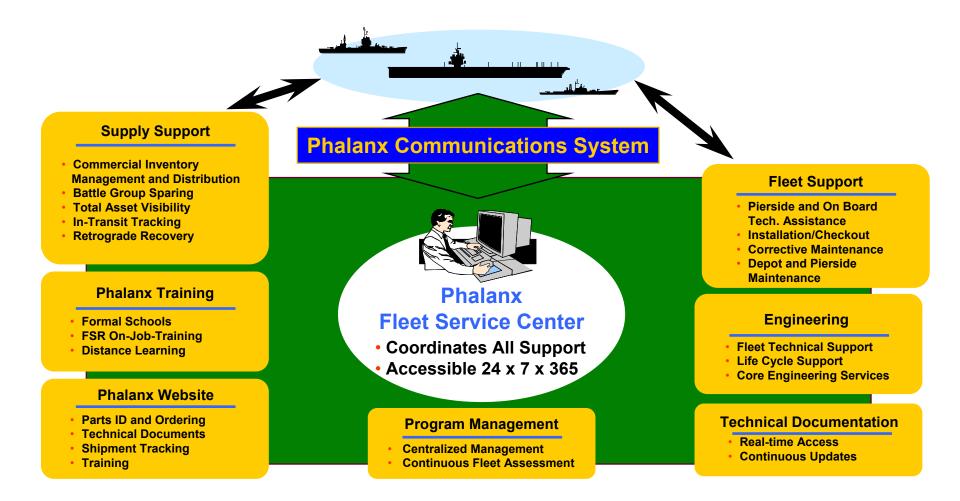
Benefits:

- Supports Multiple Roles In Ships Self Defense
- Man-in-the-Loop, Autonomous or Integrated Operation
- Fast Reaction



Full Service Contractor Phalanx Life Cycle Support



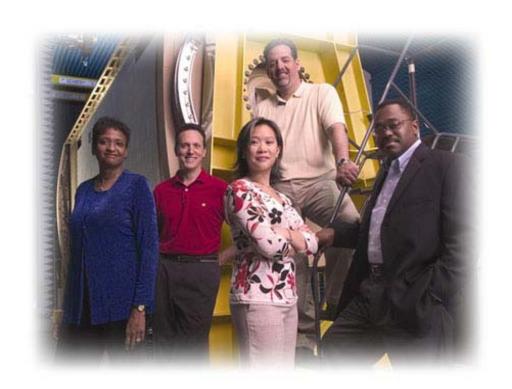


Raytheon Phalanx Life Cycle Support Provides Continuous, Worldwide, Support for Deployed and Non-Deployed Phalanx Systems



Engineering Challenges

- Global competition for talent intensifying as innovation drives job growth in engineering, science fields
- In the U.S., fewer young people earning math & science degrees
- Generational challenges
 - Aging workforce
 - Must appeal to younger workforce





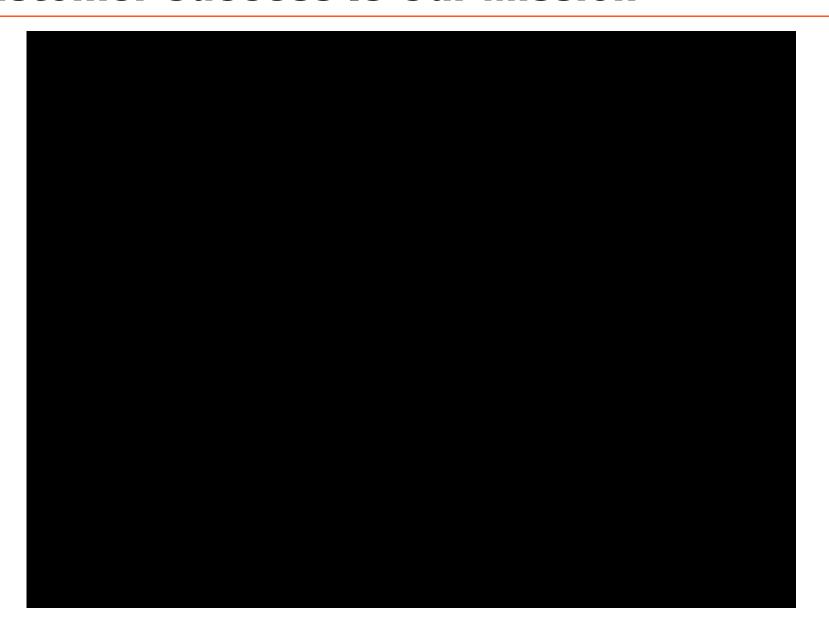
Feeding The Pipeline

- Must attract, engage diverse workforce
- Industry support/involvement in K-16 math, science education
- Partnerships with colleges, universities
 - Outstanding graduates
 - High-technology research
 - Post-graduate education
 - Creative continuing education programs
 - Outreach to the next generation





Customer Success Is Our Mission





Customer Success Is Our Mission

Customer Success Is Our Mission



The Modified Tank Ammunition **IMI M152/6 HEAT - AP - T**

National Defense Industrial Association

40th Annual Armament Systems: GARM

New Orleans, LA

April 25-28, 2005

Danny Schirding

Chief Systems Engineer

Tank Ammunition Directorate - IMI Ammunition Group

Israel Military Industries Ltd. (IMI)

P.O. Box 1044

Ramat Hasharon 47100, ISRAEL

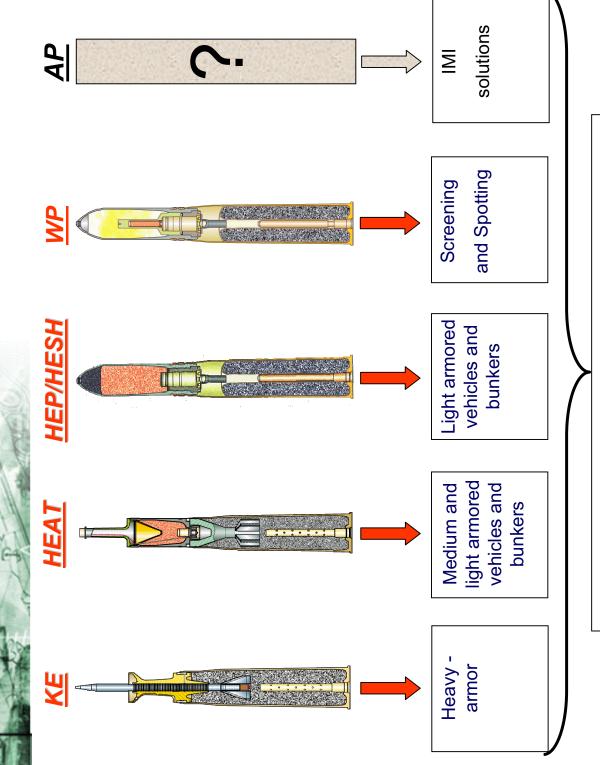
dschirding@imi-israel.com



The Main Operational Needs of **Armor Corps**

- To destroy Tanks and LAV's
- To breach and penetrate bunkers and buildings with maximum resulting damage
- To incapacitate infantry, especially AT squads.

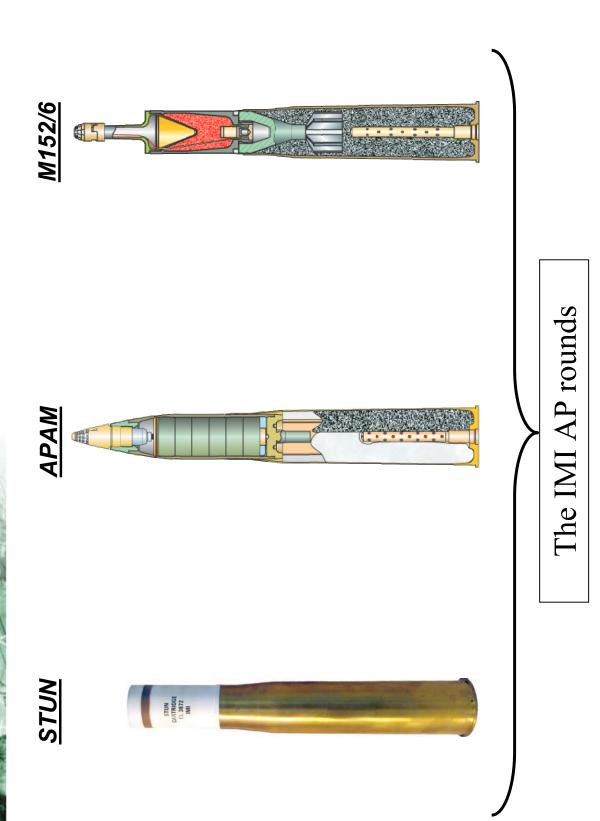




The current 105-mm family rounds









Tank Stun Rounds - Mission Statement

A less than lethal tank round for use in low intensity conflicts.

* The round is designed to deter by creating a flash, bang and blast effect similar to service ammunition.





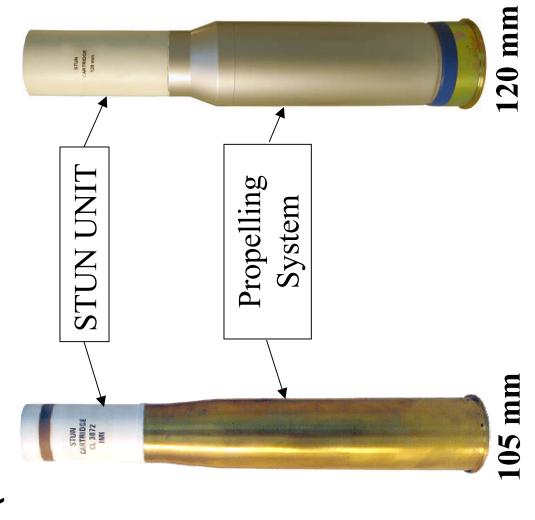
- ► Incidents involving non-combatants
- > Armed terrorists hiding behind a crowd
- ➤ Hostile civilians (mob) trying to approach/climb on the tank

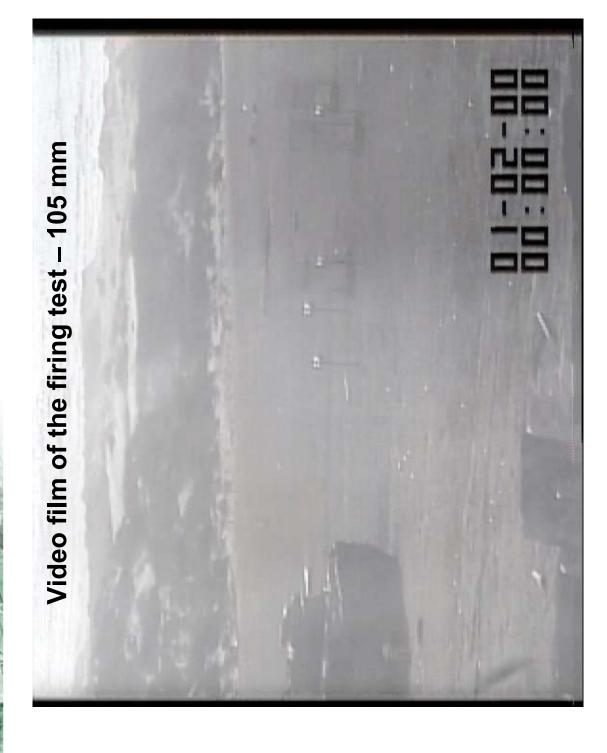




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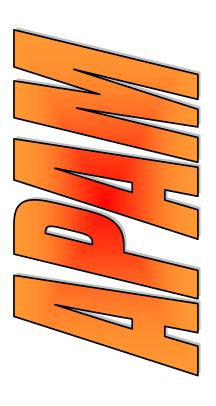
105 mm & 120 mm STUN rounds (Less-Than-Lethal tank round)











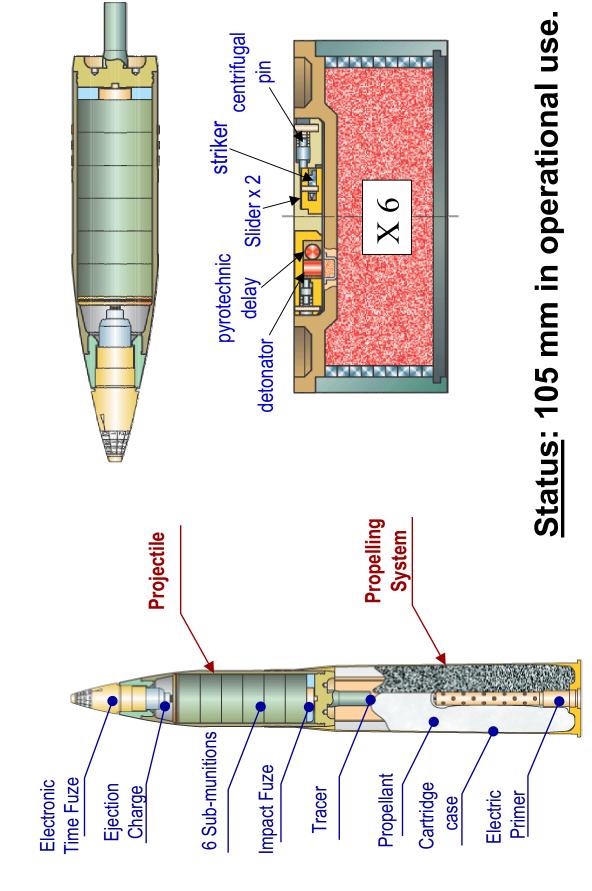


105-mm Tank Round





APAM – Anti-Personnel/Anti-Materiel

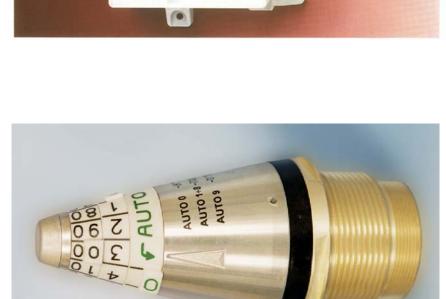


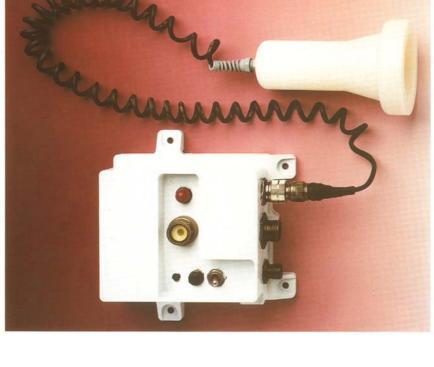


Fuze Setting

Manual fuze setting

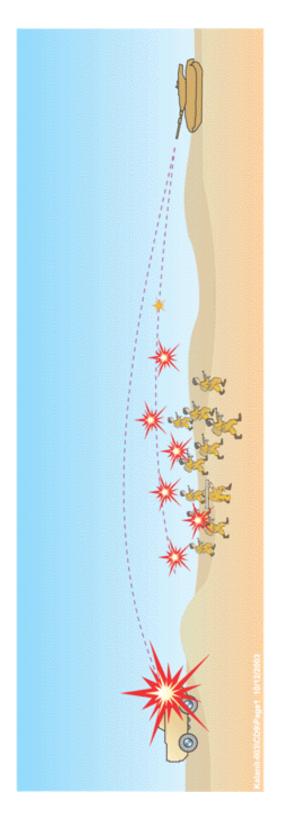








APAM – Basic Modes of Operation



Ejection Mode - Ejected sub-munitions explode sequentially in

the air after separation.

Anti-Personnel

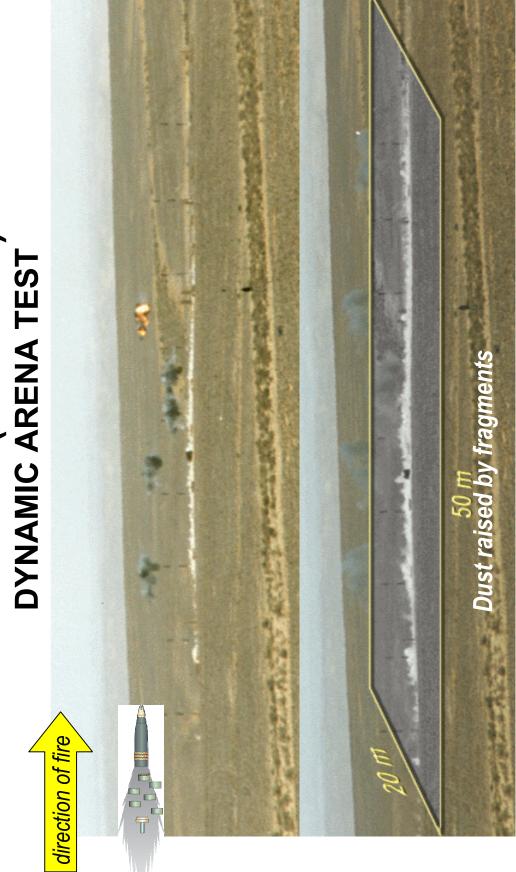
Anti-Helicopter

❖ Impact Mode – Entire projectile explodes as a unitary warhead upon impact.

√ LAV's

∀ Bunkers & Buildings

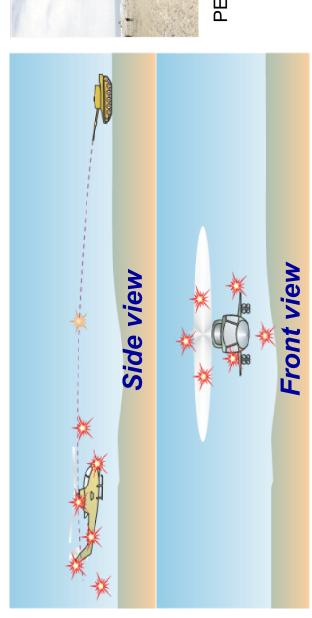
AP MODE (EJECTION)



* High effectiveness against hidden and prone targets



ANTI-HELICOPTER MODE



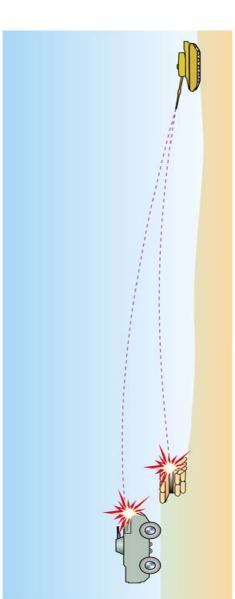


PENETRATION OF 10 mm RHA TARGET BY SUBMUNITIONS

- Six submunitions (and the projectile body & base) fly towards the target. One hit is good enough.
- Even in a near miss, the helicopter pilot will see and/or feel the detonations, causing mission abort.



AM MODE (IMPACT)







Double reinforced



Projectile will penetrate LAV's and Bunkers.

High density of lethal fragments inside.

**

concrete wall



witness plate Hits on





APAM 105 - Damage to Sand & Timber Bunker





1 ROUND



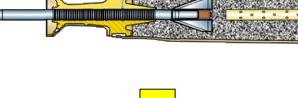
The Optimal Solution!

Infantry,

LAVs,

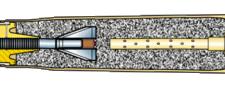
Bunkers & Buildings,

Helicopters.









- * Maximum capability with minimum rounds.
- Reduced logistic load.



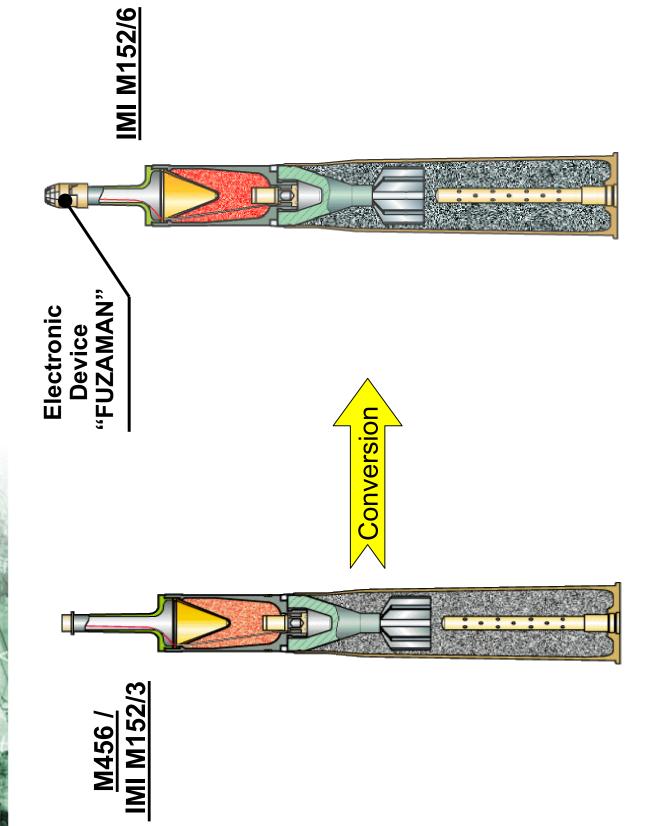
The alternative...!

Armies around the world have large stocks of 105-mm **HEAT rounds (M456 / IMI M152/3)**

IMI's alternative solution -

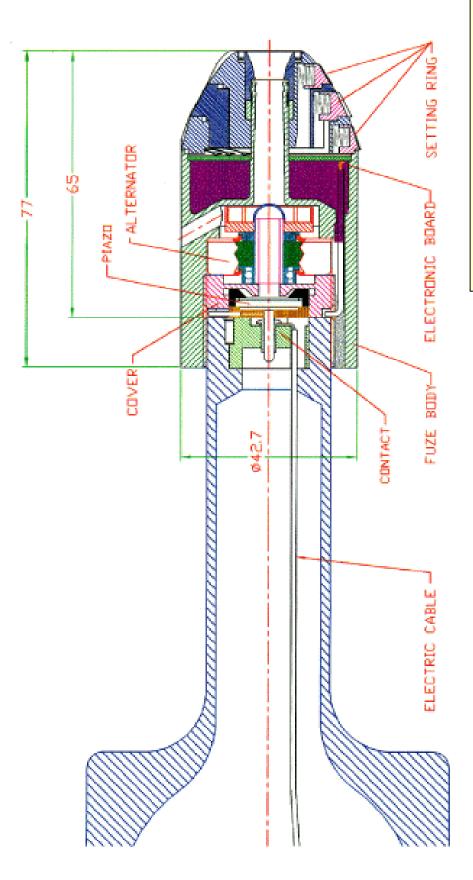
Upgrading HEAT rounds

- ➤ Using the old and well known type of ammunition
- ➤ Enhance capabilities
- ➤ Improve reliability
- ► Improve safety
- ➤ Cost effective (high kill probability)
- ➤ Providing Armor Corps needs



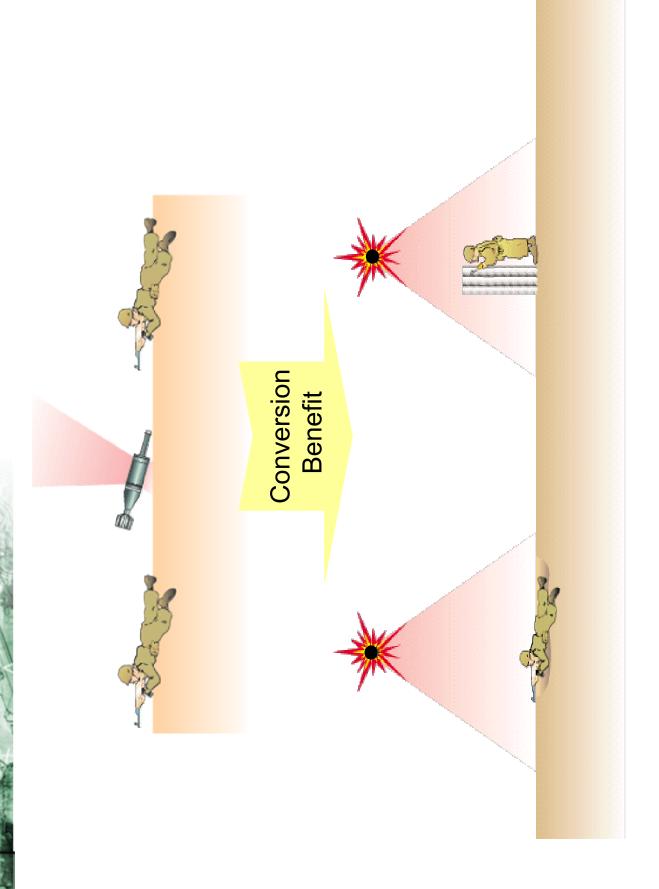
⊗: }}

High - Reliability Electronic Time Device "FUZAMAN"



RESHEF TECHNOLOGIES, LTD. AN ARYT COMPANY





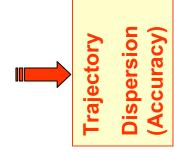


Influence on the aeroballistics performance:

- ▶ Drag Force
- ▼ Lift Force

Influence on the final ballistic

- Static and Dynamic Stability
- dunn 🖈



Penetration

s it that simple?

The operational benefits:

- ➤ Warhead detonation above the ground AP mode
- ➤ Warhead detonation upon impact and grazing (reliability and safety)
- ▼ Multi-purpose capability





Research and Development Activities

Preliminary analysis and wind tunnel tests for the

The axle of the Impeller (bar, tube)? "FUZAMAN" frequency of the alternator? The max.

Stress on the impeller?

MACH = 4.0

Number of Output

nozzle holes?

Output nozzle dia.? [mm]

Material of impeller?

Stability? Time to

Stability point? Energy at

> Height of the impeller wings? [mm]

Input nozzle dia.? [mm] 2 parallel

energy sources?

Mach and spin influence?



Aeroballistics analysis and wind tunnel tests for the Projectile of IMI M152/6

Wind tunnel tests

➤ Mach numbers: 1.2, 1.6, 2.0, 2.2, 2.6, 2.8

➤ Angle of attack: -7° ≤ α ≥ +7°

▼ Cd vs Mach

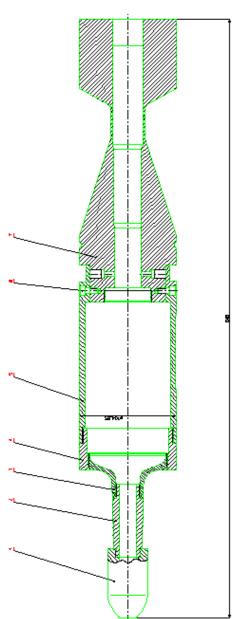
Aerodynamic coefficients (Cmα, Cnα, Croll, Clα etc.)

▼ Xcp – Xcg (static stability)



Prototypes for

Wind tunnel:



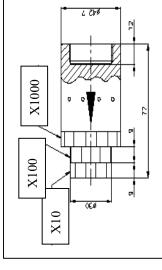
Type No. 2

X1000

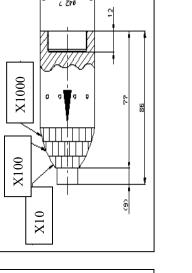
X100

X10

Type No. 1



Type No. 3



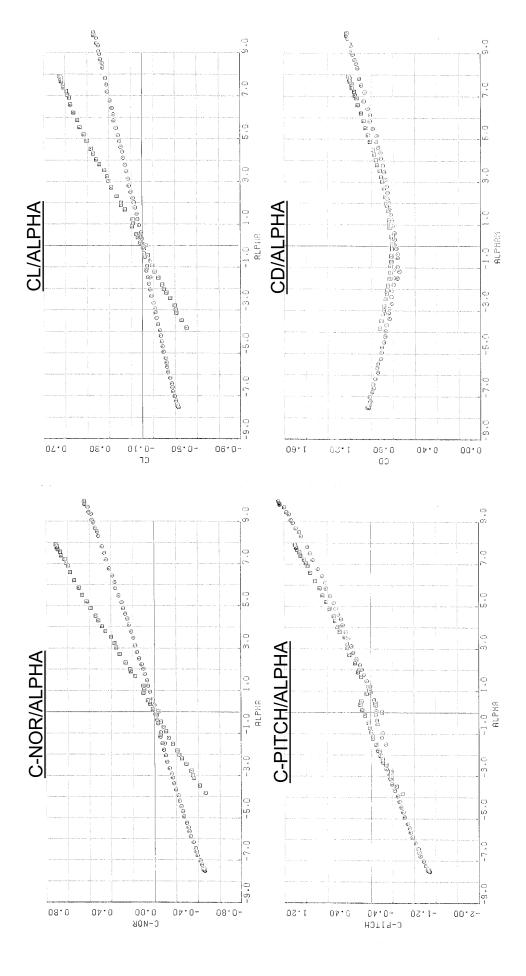


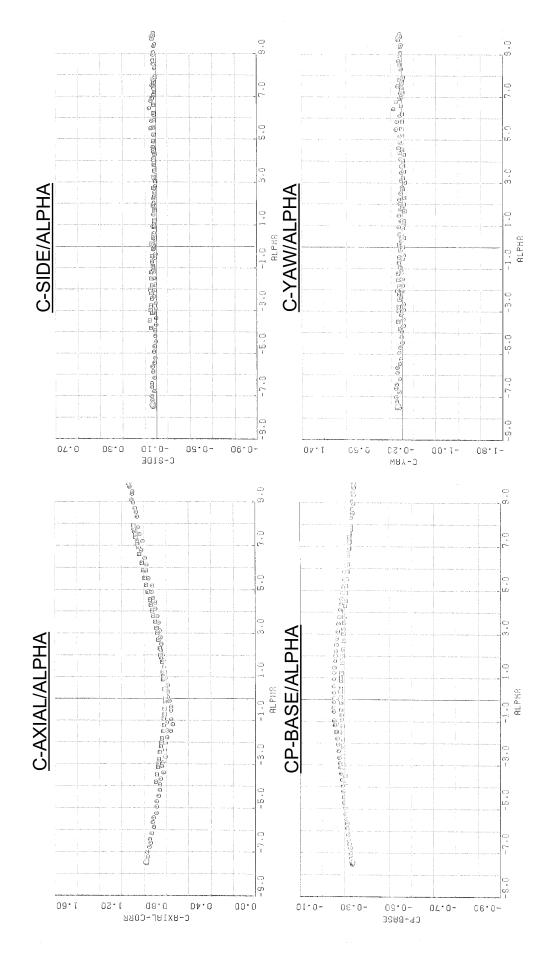


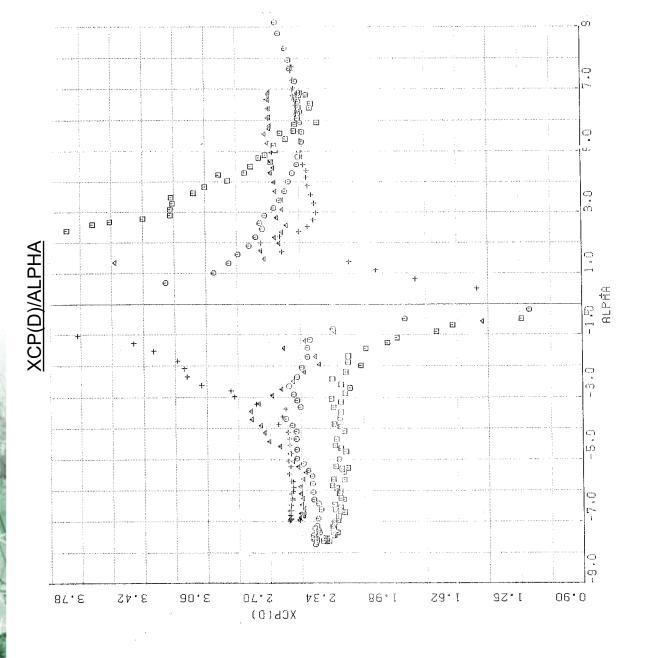










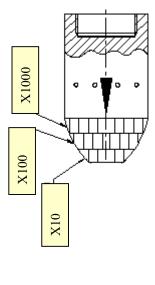


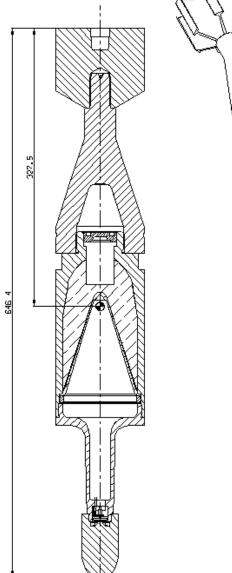


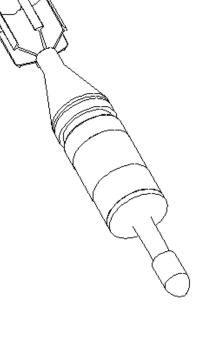
8

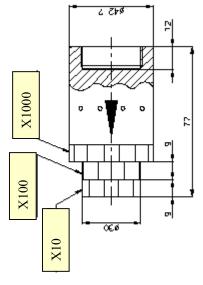
External Ballistics test - IMI M152/6

Prototype No. 1

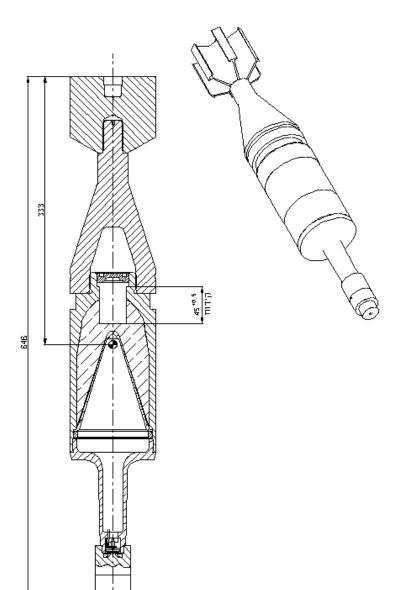




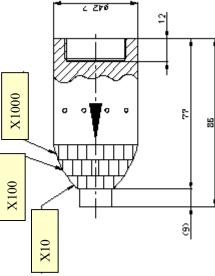




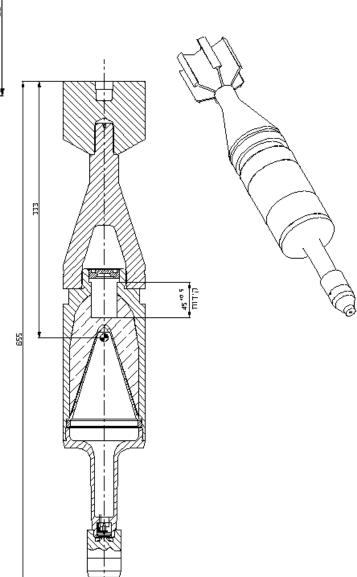
Prototype No. 2



8



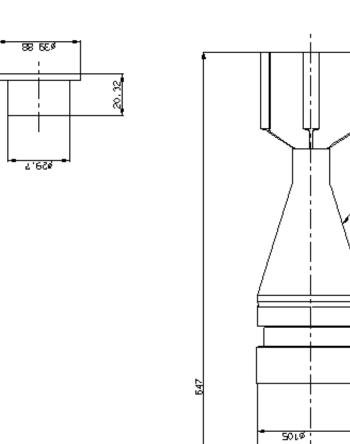
Prototype No. 3



8

M456 / IMI M152/3

(Reference)



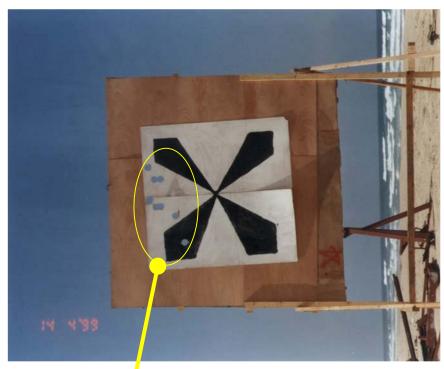
tail

body

spike



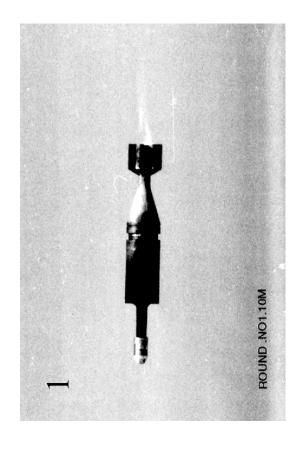
❖ Dispersion / accuracy (2,000 m)

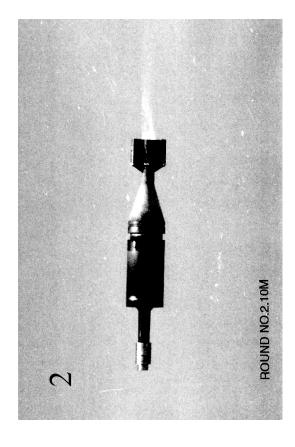


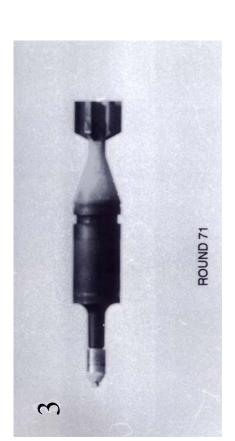


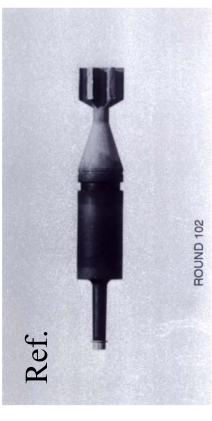


❖ Ballistically matched trajectory



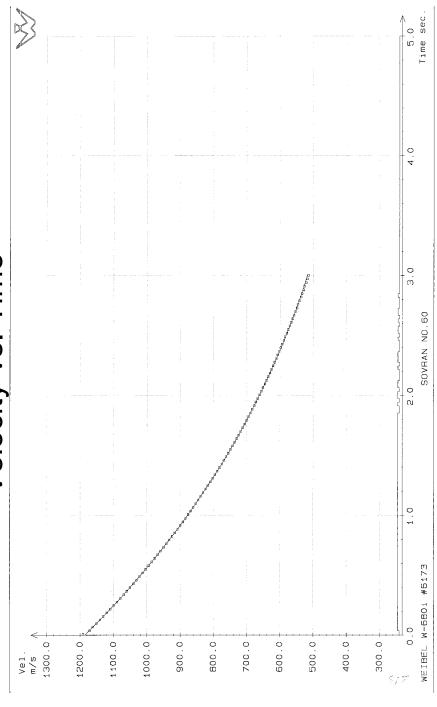








Velocity vs. Time





Final Ballistics test - IMI M152/6

Safety Firing Test

➤ Simulated cartridge with pyrotechnic (flash) composition





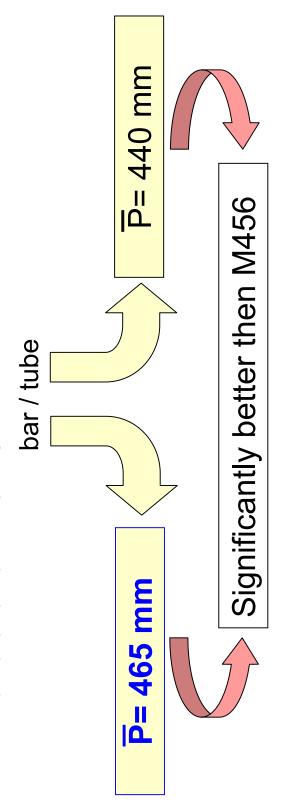


❖ Firing test – Yaw

- ➤ Wave length
- ➤ Dynamic stability

❖ Penetration tests

- ➤ M152/3 warhead
- > RHA target (225 mm plate at 120-m from the muzzle)
- V 60° NATO
- ➤ Alternator axle in the "FUZAMAN":





Front Side





Back Side



Dynamic arena test (AP mode)







* Reliability - Detonation above the ground (AP mode)





Operational Research -

➤ Lethal Area - 20x50 m

Criteria: Personnel Enemy

Standing / Prone 30" assault

Firing: 1 round / series of 3 rounds

Remaining velocity - 855 m/sec (2,000 m)

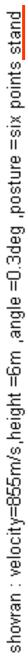
➤ Angle of fall - 0.3 deg.

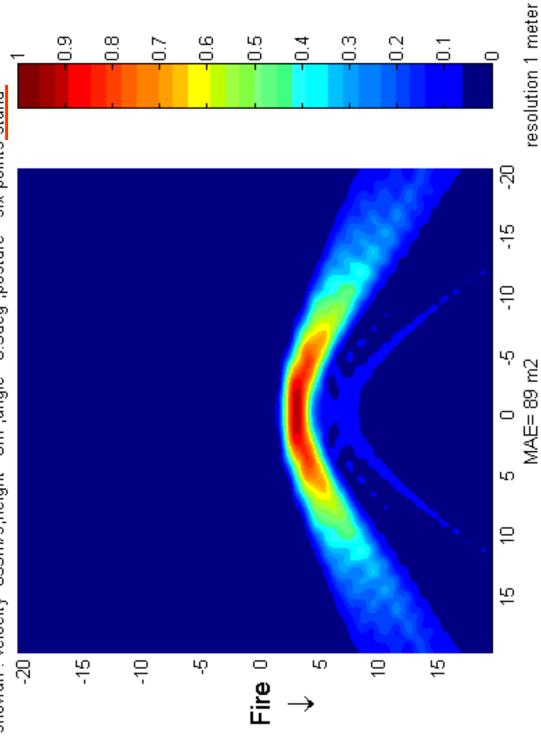
❖ Results -

- The optimal height of detonation (above ground) 6 m
- Mean Area of Effectiveness (MAE) / Lethal Area and Incapacitation Probability Maps

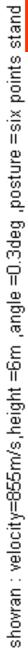


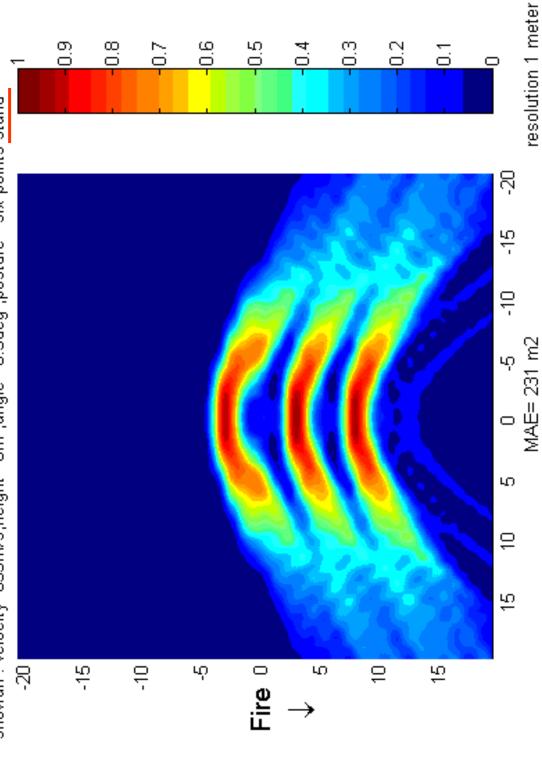




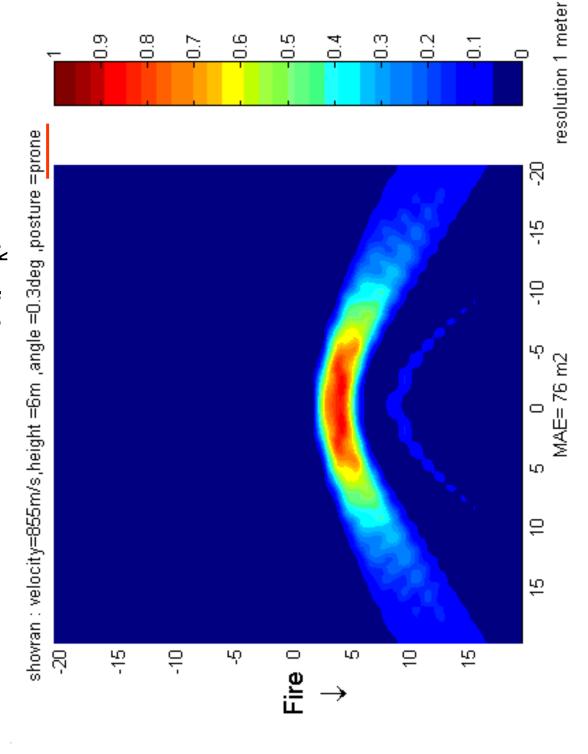




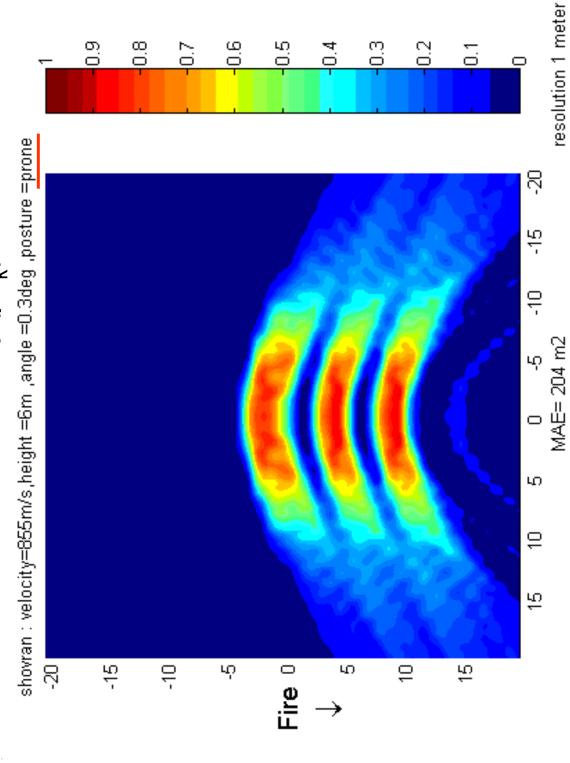












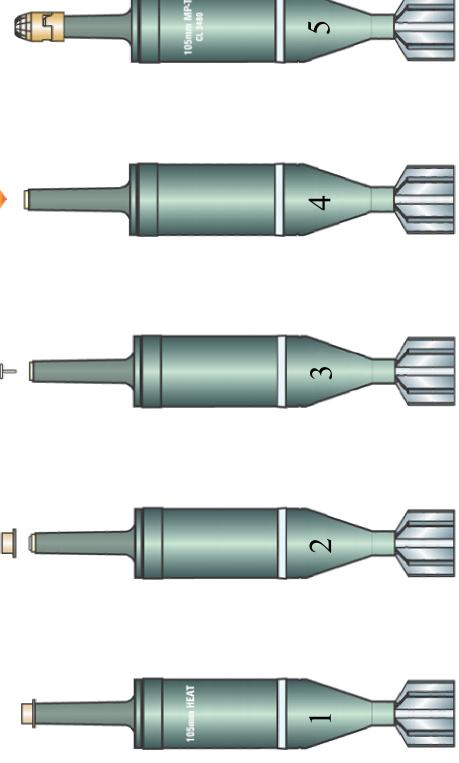


❖ Grazing (impact switch) Functioning test



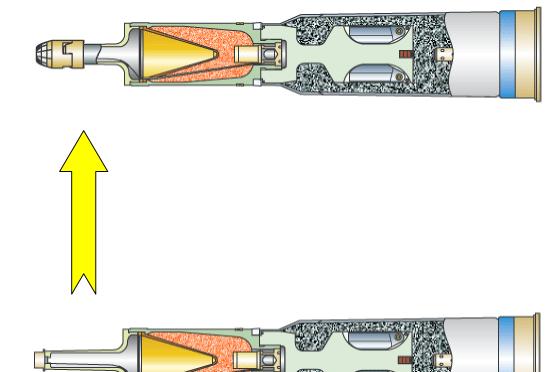


Conversion of M456 or IMI M152/3 to IMI M152/6 at field level





Growth Potential – 120 mm



CHABACTERISTICS

	25 kg 984 mm		15 kg	. 726 mm	steel	Comp. B, 1.8 kg
						Comp.
		•				
Cartridge	Weight Length	Projectile	Weight	ength	Body material	Explosive

Cartridge case combustible Propellant M30, 5.6 kg Primer electric, M4513 Fuze dual mode, electronic time/point initiated base detonation (ET-PIBD)			_		_
Iponents I electronic time/pose detonation (E1-PIBD)		ğ	3	215	JE (
Iponents I electronic time/pose detonation (E1-PIBD)		St	9	¥	‡
Iponents I electronic time/pose detonation (E1-PIBD)		园	143	2	Ξ.
Iponents I electronic time/pose detonation (E1-PIBD)		E	30	.≌	=
iponents ele ual mode, electronic time/p se detonation (ET-PIBD)		8	Ë	臣	- 등
iponents ual mode, electronic tir se detonation (ET-PIBD)			-	<u>a</u>	÷
Other Components Cartridge case Propellant Primer Fuze dual mode, electronic tir base detonation (ET-PIBD)			- 1	Ф	<u> </u>
Other Components Cartridge case Propellant Primer Fuze dual mode, electronic base detonation (ET-PIE					÷ Q
Other Components Cartridge case Propellant Primer Fuze dual mode, electron base detonation (ET-F			- 1		유민
Other Components Cartridge case Propellant Primer Fuze dual mode, electr base detonation (E		- 1	- 1	- 1	등프
Other Components Cartridge case Propellant Primer Fuze dual mode, electore					表巴
Other Components Cartridge case Propellant Primer Fuze dual mode, e			- 1		_ ≡
Other Component Cartridge case Propellant Primer Fuze dual mode, base detona	23	- 3	- 1		e.:≅
Other Compone Cartridge case Propellant Primer Fuze dual modeto					용밑
Other Compor Cartridge case Propellant Primer Gual n Fuze base de	9		- 1		혈읈
Other Compo	5				E e
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Other Co Cartridge cas Propellant Primer Fuze	Ξ		- 1		년 %
Other C Cartridge c Propellant. Primer Fuze	0	33			
Other Cartridge Propellan Primer Fuze	C	Ö	نب	- 1	
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Cart Prop Prim Prim Fuze	9	금	9	EL	
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	0	చ	풉	풉	团

BALLISTIC PERFORMANCE

≥ .	Muzzle velocity	L C	1078 m/s	78 m/	- CO E
⋖	Accuracy, typical SD	C7.0	U.Z.5 mill, H and E	and	ш
Ш	Effective range mo	more than 3000 meters	3000	meter	20
S)	-	projectile detonates 5 m	rtonate	38 5 T	E
_		inoub e	P		

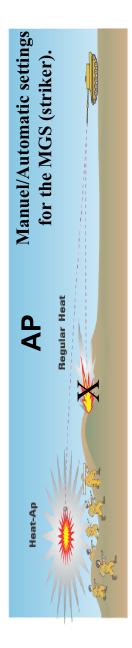
BALLISTIC PERFORMANCE

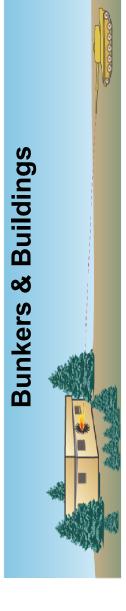
-40 to +52°C	-40 to +63°C	standards
-40	-40	and NATO
	Temp. limits, storage	Various tests IAW MIL-STD-810D and NATO standards
firing	storage	IAW MI
Temp. limits, firing	emp. limits,	arious tests
		parameter .

a: {

Summary - Targets and Operating Modes

MBT & LAV









Overview

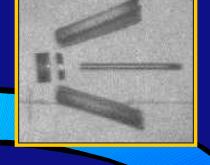
Propellant

Flight









IM



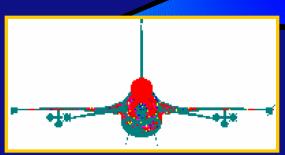
Lifetime prediction

Warhead



Surveillance

Performance

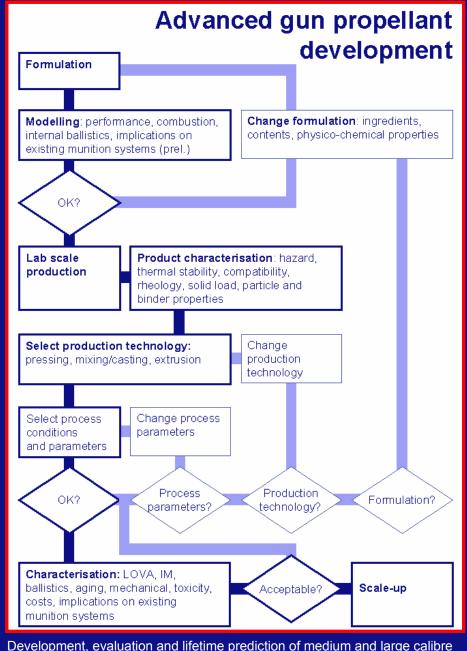


Effectiveness









Propellant: Capabilities

- Modeling & simulation
 - Thermodynamics
 - Processing
 - Internal ballistics
- Lab-scale production
 - Up to ~ 1 kg (analyses)
- 'Small scale' production
 - Up to ~ 300 kg
- Performance testing
 - Closed & vented bombs
 - Test guns
 - Thermal, IM & safety properties

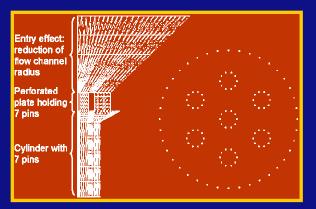


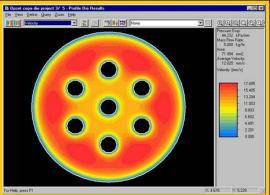
ammunition

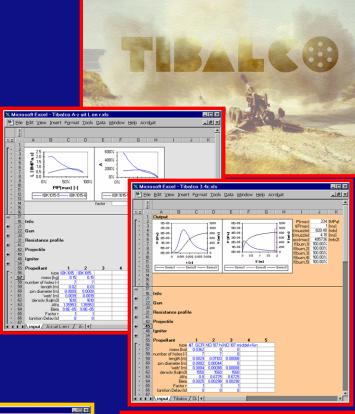
3

Propellant: Modeling & simulation

- Thermodynamics
 - NASA-Lewis, Blake, ICT-code
- Internal ballistics
 - TIBALCO (TNO Internal BALlistic Code)
- Processing
 - Rheology
 - Extrusion & shaping processes



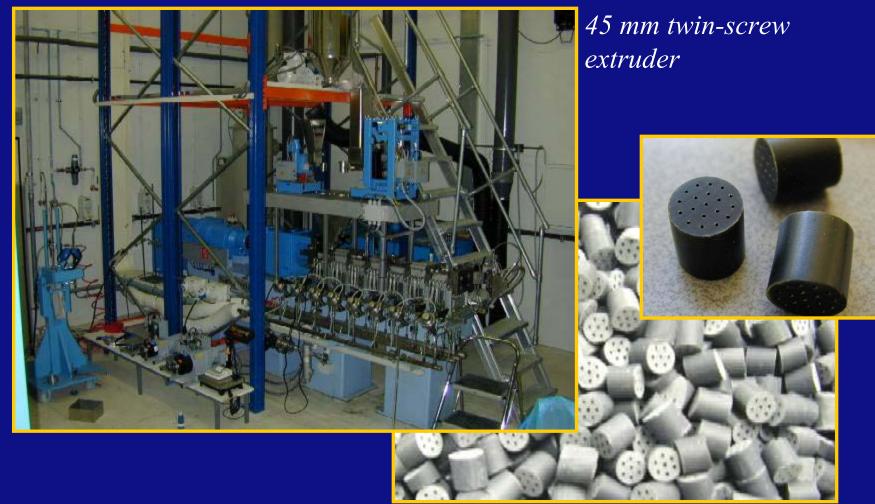








Propellant: Processing



Gert Scholtes, 40th GARM, April 2005

Propellant: Test facilities

- Closed Vessels
 - 43.5 cc / 130 cc LPCV (20 MPa)
 - 25 700 cc CV (150 500 MPa)
 - 400 cc HPCV (1000 MPa)
- Erosivity & burning interruption tests
 - 130 cc 20 MPa
 - 500 cc 150 MPa
- Plasma ignition
- Instrumented guns
 - .50 gun
 - 29-mm / 50-mm / 78-mm accelerator







Vented HPCV and catch tank



Propellant: Examples of R&D projects

- Propelling charge development
- Temperature independent propellant
- Barrel erosion
- Ageing & lifetime assessment



Stick propelling charges for excellent ignition behaviour



960 940 920 velocity 900 880 860 840 820 800 30 -50 -30 -10 50 temperature a% • b% 50% a% + 50% no Poly. (a%) Poly. (b%) Poly. (50% a% + 50% no) Poly. (no)

Proven temperature independency



Burning properties and mechanical integrity of aged propellants

Gert Scholtes, 40th GARM, April 2005

Overview

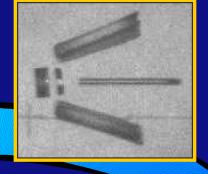
Propellant

Flight



Ignition propellant







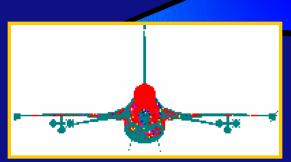
Lifetime prediction

Surveillance

Performance

IM Warhead





Effectiveness



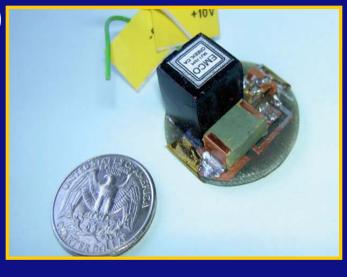
Gert Scholtes, 40th GARM, April 2005



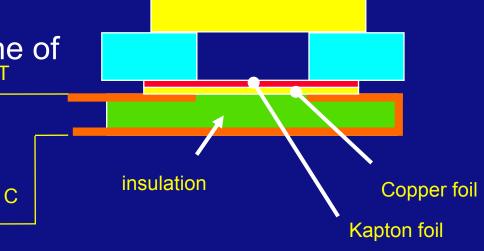
MEMs Exploding Foil Initiator (EFI)

- Intrinsic safe
 - No primary explosives
 - Not sensitive to EM fields
- Precision timing for initiation (e.g. aimable warheads)
- Very reliable

No need for out-of line of charge



Explosive



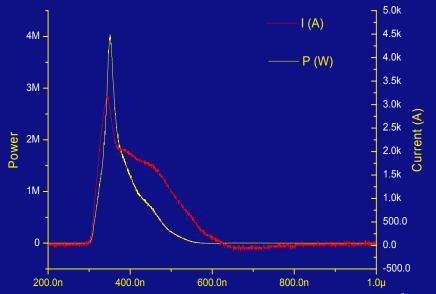


MEMs EFI: What you need

- Proper circuit with COTS components
 - Small high voltage power supply (several kV and kA)
 - Solid state Switching device
- Appropriate dimensions en properties of:
 - Exploding foil
 - Flyer plate
 - Strip-line
 - Barrel
- Pressed HNS-IV crystals at the right density

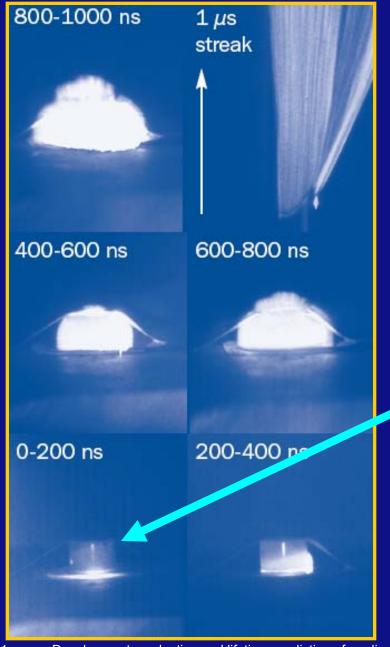


Performance of an optimised EFI-circuit

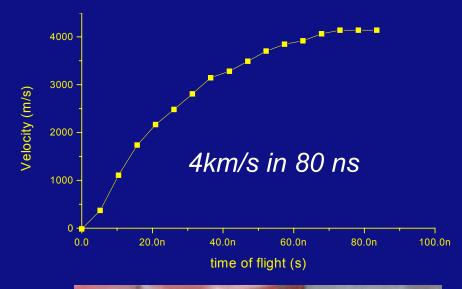


Gert Scholtes, 40th GARM, April 2005





Ignition train: MEMs EFI









Overview

Propellant

Flight











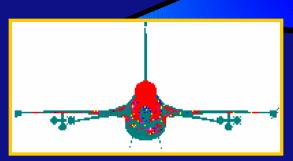
Lifetime prediction

IM Warhead



Surveillance

Performance



Effectiveness

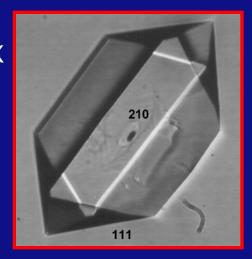






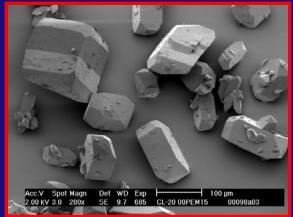
Warhead: recrystallisation to obtain the next generation of explosives

RS-RDX



Insensitive crystals for HE Warheads

CL-20



HNF



Insensitive crystals for rocket propellants



HNS-IV

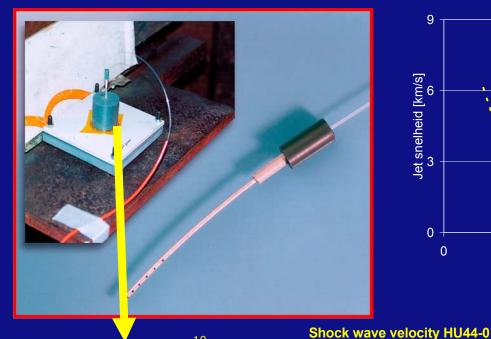
Insensitive crystals for Booster

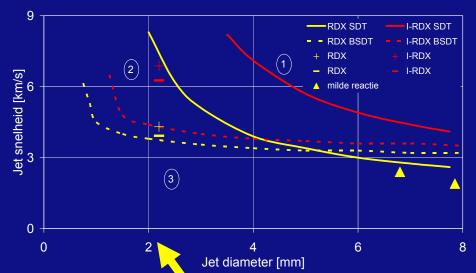
Explosives

Gert Scholtes, 40th GARM, April 2005



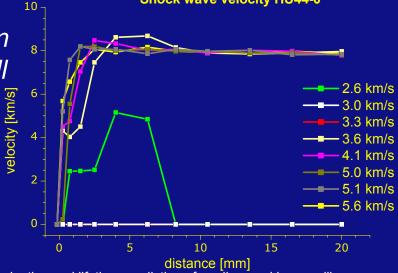
Warhead: characterisation of explosives

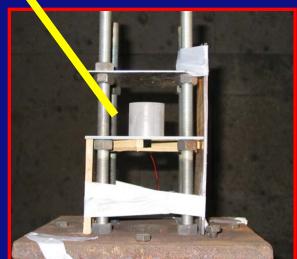




Shaped charge testing and simulation (PBXN109)

Shock initiation testing of small samples





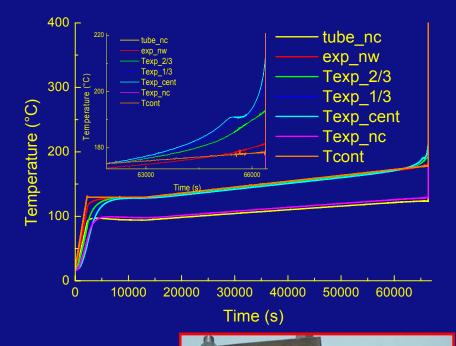
Gert Scholtes, 40th GARM, April 2005

Warhead: Understanding the behaviour of

explosives and IM









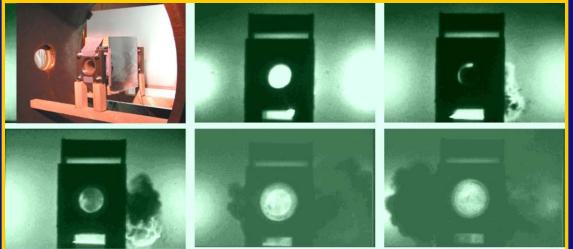
Cook-off testing and Simulation



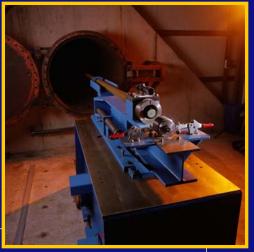


Warhead: Understanding the behaviour of

explosives and IM

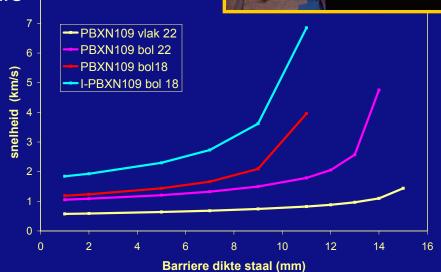


Bullet/Fragment testing and simulation



The responses of a confined materials after the impact of a fragment.





Overview

Ignition

propellant-

Propellant

Flight







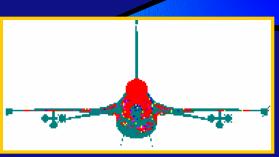
Lifetime prediction

Surveillance

Performance

IM Warhead





Effectiveness

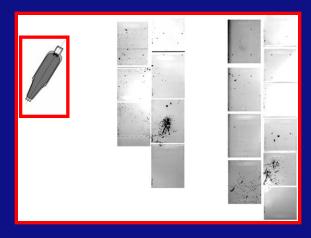


Hit Gert Scholtes, 40th GARM, April 2005

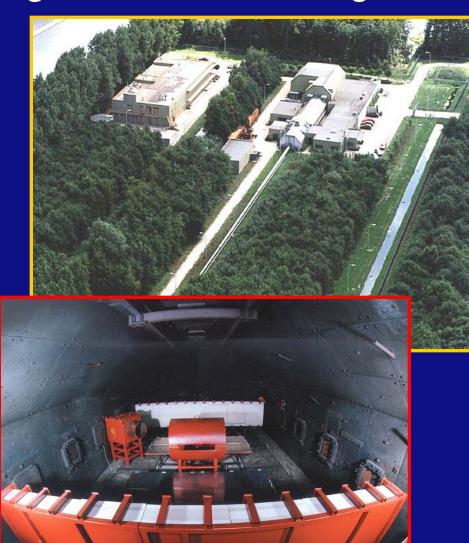


Effectiveness: Fragmenting ammunition testing

- 60 m range for HE ≤ 76 mm
- 200 m range for KE ≤ 40 mm
- Bunker for ≤ 25 kg TNT



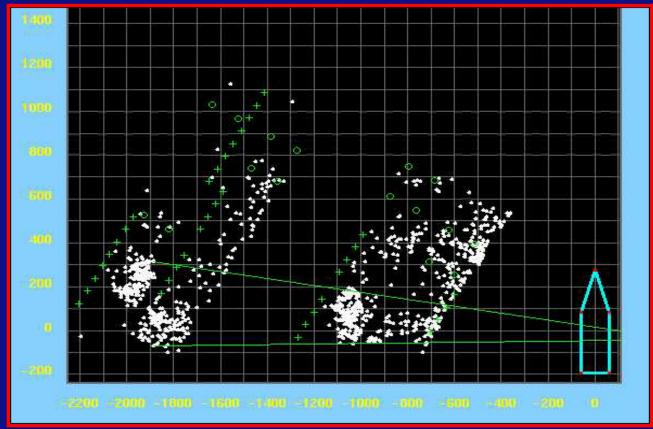
- Fragment cloud analysis method
 - Rotational symmetry
 - Cylinder with windows
 - Cardboard soft recovery
 - X-ray shadowgraphs





Effectiveness: Fragmenting ammunition testing

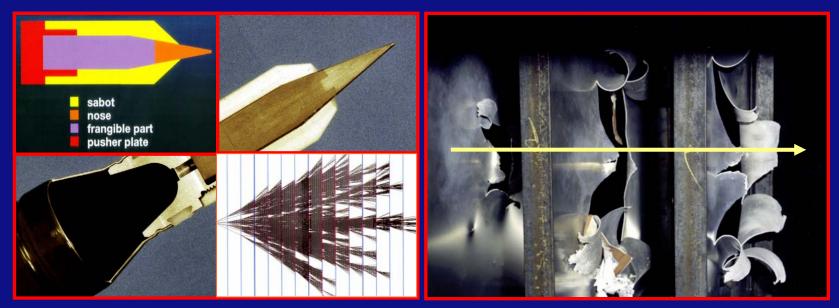
- Fragment distribution
 - Spatial
 - Velocity
 - Mass
 - Energy



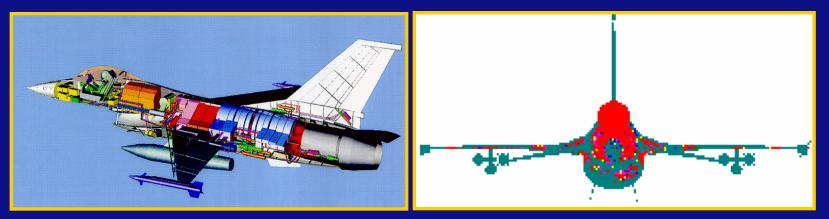


Gert Scholtes, 40th GARM, April 2005

Effectiveness: Munition Lethality/Platform Vulnerability



Terminal ballistics experiments & simulations



Lethality / vulnerability simulations



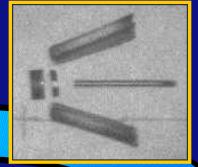
Overview

Propellant

Flight







Ignition train



Lifetime prediction

Surveillance

Performance

IM Warhead





Effectiveness



Hit Gert Scholtes, 40th GARM, April 2005







ammunition

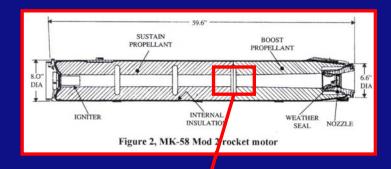
Lifetime prediction: Ageing of missile

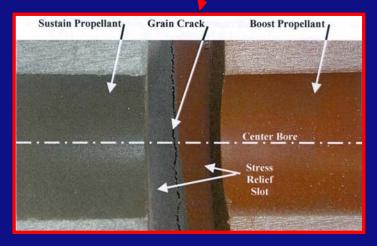
US AIM-7 Sparrow incidents (1997 & 1999)





US MK-58 Mod 2 motor investigation

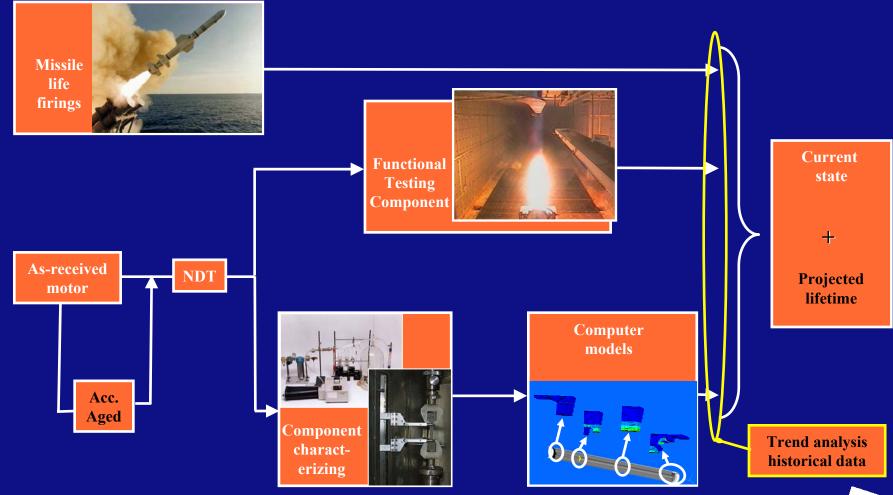




Source: paper P. Huisveld AVT-RTO-089, 2002 Aalborg



Lifetime prediction: Element "toolbox" for missiles





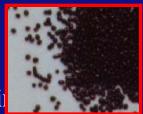
Surveillance of gun propellants







Range of 5 sample vessels covers the whole range of propellant grains



No pre-treatment of grain necessary

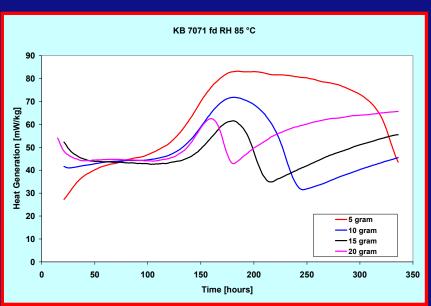






Surveillance of gun propellants

- Heat flow Calorimetry (HFC) with full size grains
- Heat generation in time as function of loading density of vessel
- →Munition like testing







ammunition

Lifetime production and surveillance: Products

- <u>Lifetime studies (Toolbox)</u>
- <u>Surveillance methodology</u> for gun propellants (realistic comparison to ammunition situation, including
 - Equipment
 - Tailor made training programme
 - Tailor made munition management system
 - Guarantee and spare parts



Summary

- TNO Defence, security and safety is an independent organisation and a strategic partner for the Dutch Ministry of defence
- We also use our accumulated expertise for foreign governments and for defence related industries.
- R&D → development → prototyping → pre-production →
 production → in service, of munition: TNO has the expertise for
 Effective and Insensitive Munitions development.
- But also the expertise for lifetime predictions and surveillance of propellants.
- Combination of experimental facilities, theoretical knowledge/expertise and model/computer codes makes TNO a qualified partner for your future munitions development.



Gert Scholtes, 40th GARM, April 2008



LINE OF SIGHT/BEYOND LINE OF SIGHT (LOS/BLOS) ADVANCED TECHNOLOGY DEMONSTRATOR (ATD)

BRIEFING FOR THE GUNS, AMMUNITION, ROCKETS & MISSILES SYMPOSIUM - 25-29 APRIL 2005

Providing America Advanced Armaments for Peace and War



DAVID C. SMITH, P.E.
ANTHONY J. CANNONE
LOS/BLOS ARMAMENT TEAM



LIGHTWEIGHT 120 MM GUN (LW120) LOS/BLOS ATD



- OVERVIEW OF LOS/BLOS ATD GUN
 - DESIGN CONCEPT
 - REQUIREMENTS
- OVERVIEW OF TEST STATUS TO DATE
 - TEST RESULTS PHASE 1
 - TEST RESULTS PHASE 2
 - TEST RESULTS PHASE 3 (PRELIMINARY TEST DATA)
 - IN HOUSE TESTING
- PATH FORWARD
- ACCOMPLISHMENTS



LOS/BLOS ATD GUN OVERVIEW

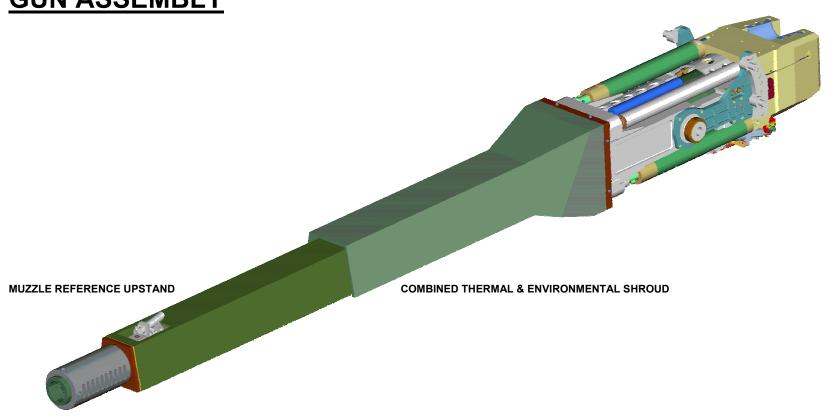


- GOAL: DEVELOP A LIGHTWEIGHT 120MM GUN FOR USE ON A 20 TON VEHICLE
 - WEIGHT: < 4,400 POUNDS</p>
 - RECOIL IMPULSE: < 5,300 POUND-SECONDS</p>
 - RECOIL LENGTH 25 INCHES
 - SPACE CLAIM LIMITS OF MCS CONCEPT VEHICLE
- STATUS & ONGOING ACTIVITIES
 - VEHICLE DYNAMIC RESPONSE DEMONSTRATOR (VDRD) GUN DEVELOPED AND COMPLETED TESTING IN FEBRUARY 2004
 - LINE OF SIGHT/BEYOND LINE OF SIGHT (LOS/BLOS) ADVANCED TECHNOLOGY DEMONSTRATOR (ATD) PROGRAM BUILT AND TESTED 3 VERSIONS OF THE LIGHTWEIGHT 120MM GUN FROM NOVEMBER - MARCH 2005
 - KEY TECHNOLOGIES CONTINUE MATURING IN NEW PROGRAM





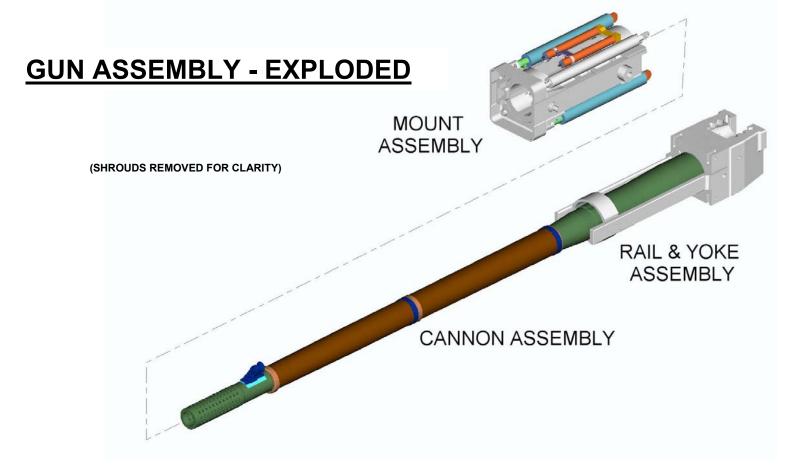
GUN ASSEMBLY



ATD 1 BLAST DEFLECTOR SHOWN THIS VIEW



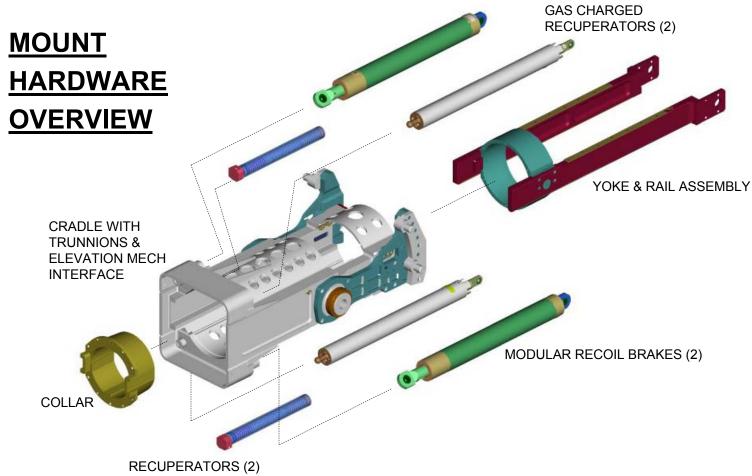






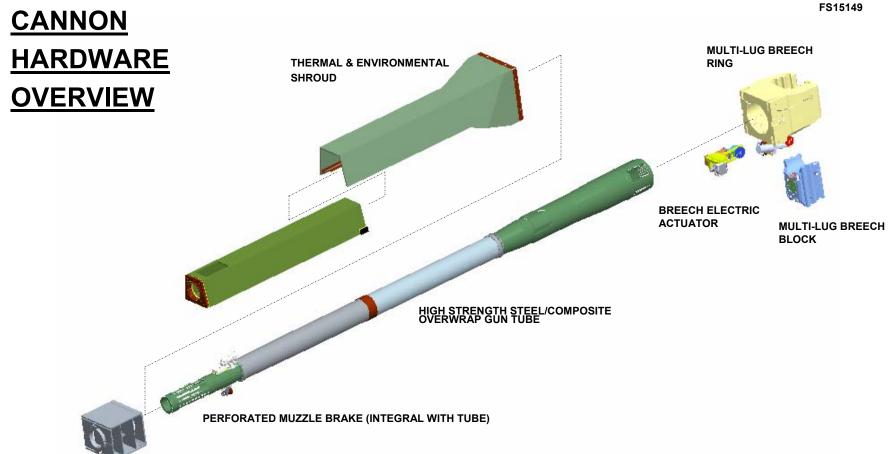


FS15149









BLAST DEFLECTOR (ATD 2 VERSION SHOWN)



LOS/BLOS ATD GUN DESIGN CONCEPT (ATD # III.WP.2003.01)



	Current	LOS/BLOS ATD		
Capability	Capability (Baseline)	Minimum	Goal	
Armament				
 Gun weight (lbs) 	6700 lbs	4,400 Lbs	4000 Lbs	
 Gun elevation 	20 ⁰	30 ⁰	50°	
 Recoil & Ammo Volume 	4.5 m ³ (Abrams)	3.5 m ³	3.5 m ³	
 Stowed Rounds 	42 (Abrams)	38	43	
 Weapon Recoil Force 	160,000 Lbf	90,000 Lbf	85,000 Lbf	
 Weapon position error 	.75 mil (El and Az)	.5 mil (El and Az)	.35 mil (El and Az)	
(LOS) • Ammunition	(M1) dynamic	dynamic	dynamic	
 Advanced KE Armor Penetration 	M829A2 @ 2km	Advanced Armor Threat @2km	Advanced Armor Threat @4km	
• MRM – PSSK • Warheads:	N/A	Pssk (class) 2-12km	Pssk (class) 2-16km	
Warhead – SC L/D (pen) Capability	1.7 Armor	1.0 (class) Armor	1.0 (class + 30%) Armor, personnel Bunker, Urban	
Precision Ignition Demo	T2 s of 3 ms (abrams)	T2 s of 0.5 ms	Helicopter T2 s of 0.1 ms	

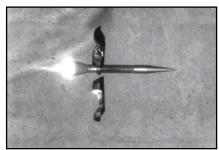






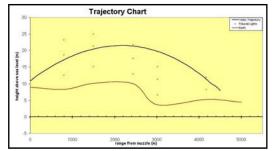
- TEST SET UP AT ATC TRENCH WARFARE II FIRING RANGE
- MULTIPLE SOFT TARGET RANGE UP TO 5000M LOS, VIDEO SCORING
- 2 SEPARATE LOF ALLOW DU AND HEAT TO BE FIRED
- FIRE CONTROL DATA COLLECTION
- HIGH SPEED PHOTOGRAPHY, MUZZLE EXIT BALLISTICS
- NEAR REAL-TIME PROCESSING USING TEST SITE INTEGRATION
- RANGE ALLOWED MRM FIRING TO 12 KM

















- ATD GUN FIRING TEST RESULTS TO DATE
 - COMPOSITE JACKETED GUN TUBE FIRED & STRAIN MEASURED
 - STRUCTURAL INTEGRITY DEMONSTRATED, PERFORMING WELL.
 - TRUNNION FORCE MEASUREMENTS:
 - PRELIMINARY RESULTS INDICATE GOOD AGREEMENT WITH PREDICTIONS
 - PEAK FORCES FIRING M829A3 WITH MUZZLE BRAKE < 65,000 LBS
 - MUZZLE BRAKE EFFICIENCY GOAL OF > 25% FIRING THE M829A3 ROUND DEMONSTRATED.
 - MRM-KE ROUND FIRED TO MAX RANGE (30° ELEVATION)
 - 12 KM RANGE DEMONSTRATED, ACTUAL RANGE 12.8 KM.
 - PROPER FIN DEPLOYMENT THROUGH MUZZLE BRAKE DEMONSTRATED.





- RESULTS (cont'd)
 - INITIAL EROSION TESTING ON CYLINDRICAL MAGNETON SPUTTER (CMS) COATED (TANTULUM) LINER (~2')
 - ACCURACY TESTING TARGET IMPACT DISPERSION (TID)
 - M829A3 INITIAL TID COMPLETED
 - M865 & M831A1'S GOOD RESULTS TID WITHIN REQUIREMENTS.
 - BREECH BLOCK WITH AMMO DATA LINK (ADL).
 - FUNCTIONALITY, STRUCTURAL INTEGRITY OF DUAL DATA PIN ADL DEMONSTRATED.
 - SINGLE PIN ADL (GERMAN) SYSTEM DEMONSTRATED.
 - PRECISION IGNITION DEMONSTRATION.
 - BREECH ACTUATOR HARDWARE DEMONSTRATED
 - ULTRA HIGH STRENGTH STEEL TUBE FIRED









LIGHTWEIGHT 120 MM GUN ASSEMBLY AT ABERDEEN TEST CENTER – 19 OCT 04. NOTE MICROPHONES TO RECORD BLAST OVERPRESSURE, ADDITIONAL RISERS UNDER GUN 'KNEES' TO OBTAIN MORE ACCURATE GROUND REFLECTION DATA.

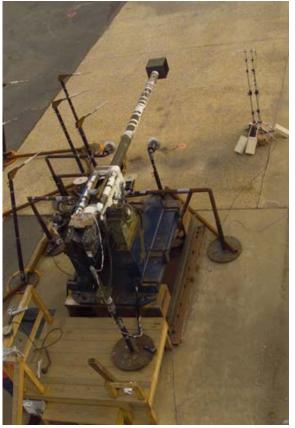
PENCIL PROBE AND HULL MOUNT SIMULATOR MICROPHONES WERE CUSTOM DESIGNED AND FABRICATED BY ATC BALLISTICS AT THE REQUEST OF THE BENET LABORATORIES







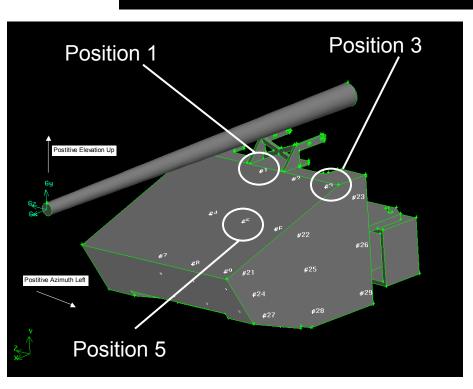


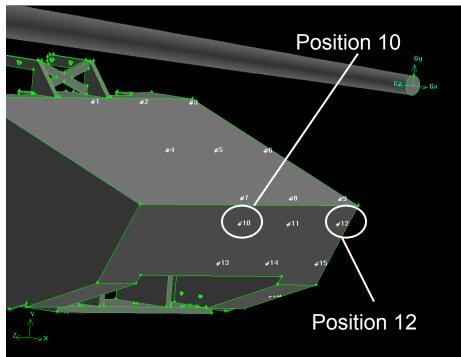






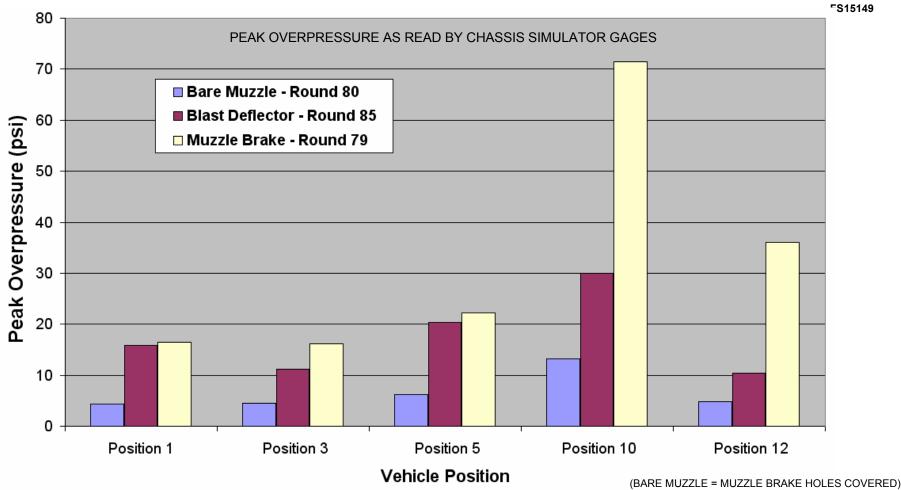
HULL POSITIONS MEASURED WITH CHASSIS SIMULATOR GAGES







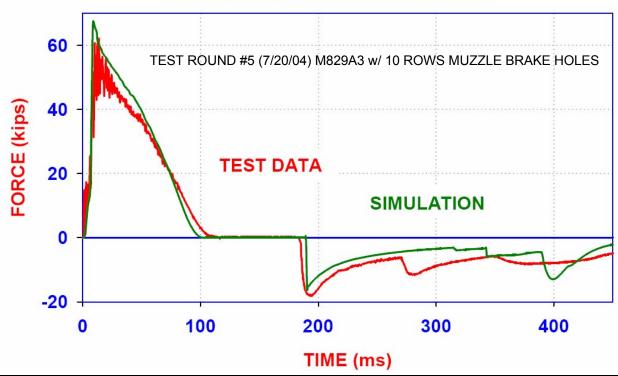








TOTAL BRAKE FORCE vs TIME

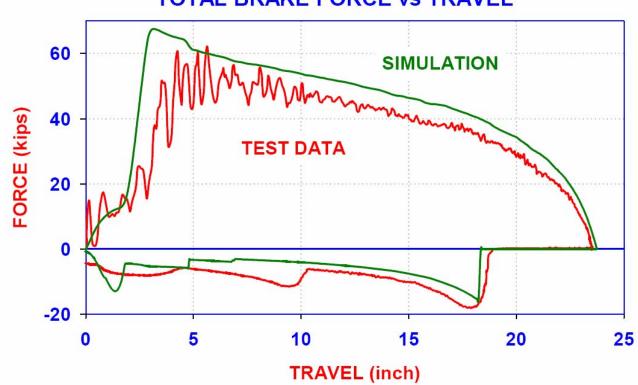


ROUND	CARTRIDGE	IMPULSE (LBSEC)	CONFIGURATION
5	M829A3	5,232	Full Muzzle Brake
6	M829A3	5,315	Full Muzzle Brake
7	M829A3	5,360	Full Muzzle Brake









ATD #1 RECOIL SYSTEM TEST RESPONSETEST ROUND #3 (7/20/04) M829A2 w/ 10 ROWS MUZZLE BRAKE HOLES



LOS/BLOS ATD GUN PRELIMINARY TEST RESULTS



ROUND M829A2 @ 70DegF

ATD - TEST	BALLISTIC IMPULSE	IMPULSE at TRUNNION	PERCENT REDUCTION
1 (3)	6428	4818	25.0
2 (10)	6327	4803	24.1

ROUND M831A1

ATD - TEST	BALLISTIC IMPULSE	IMPULSE at TRUNNION	PERCENT REDUCTION
1 (3)	5645	4322	23.4
2 (10)	5560	4442	20.1

ROUND M865

ATD - TEST	BALLISTIC IMPULSE	IMPULSE at TRUNNION	PERCENT REDUCTION
1 (3)	4833	3628	24.9
2 (10)	4825	3549	26.4











ATD MOCK GLACIS ASSEMBLED AT ABERDEEN TEST CENTER





LOS/BLOS ATD GUN TEST RESULTS





ATD 3 TITANIUM RAILS, YOKE & ADAPTOR









ENVIRONMENTAL/SIGNATURE SHROUD COMPOSITE SECTIONS (POST CURE) MADE AT BENET

ENVIRONMENTAL/SIGNATURE SHROUD ASSEMBLIES BEING TEST FIT TO TUBE AT BENET



LOS/BLOS ATD GUN IN HOUSE TESTING



COMPOSITE TUBE (ATD 3)
STRAIN TESTING AT
BENET

HIGH TENSION COMPOSITE SPECIMEN

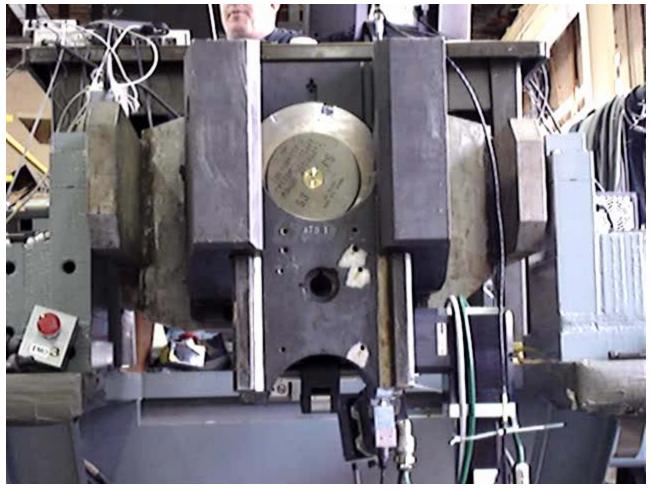






LOS/BLOS ATD GUN IN HOUSE TESTING







LOS/BLOS ATD GUN PATH FORWARD



- COMPLETE ANALYSIS OF ALL TEST DATA OF LOS/BLOS ATD TEST PHASES (ONGOING THROUGH MAY 05) AND INCORPORATE INTO GUN TECHNICAL DATA PACKAGE
- START NEW STO PROGRAM "LIGHTWEIGHT ARMAMENT ENHANCEMENTS"
 - EXPLORE/MATURE HIGHER RISK ARMAMENT WEIGHT REDUCTIONS AND ACCURACY ENHANCMENTS
 - DEMONSTRATE TECHNOLOGIES FOR TRANSITION INTO SDD
- SDD PROGRAM START
 - TAILOR GUN INTERFACES FOR THE MCS
 - REFINE TECHNICAL DATA PACKAGE TO OPTIMIZE COST & PERFORMANCE
 - INCORPORATE INTERFACES FOR NEW MUNITIONS



FCS/MCS LW120 GUN DESIGN CONCEPT – GUN ASSY WEIGHT



Total Weight (lbs)	6,665.00	4,443.00	3,630.00		
Bore Evacuator	50	N/A	N/A		-115 LBS – ATD DESIG DEMO IN LAEP
Rails, Yoke, Adapter	(included)	283	168	 	
Recuperators (2)	(included)	84	75		-9 LBS - RECUPS
Recoil Brakes (2)	(included)	164	100	K	-64 LBS - BRAKES
Elevation interface plates	(included)	78	78		
Replenisher Assembly	(included)	18	18		
Cradle Assembly	2560	532	532		DEFLECTOR
Blast Deflector	N/A	90	65	\	-25 LBS – BLAST
MRS	0	0	0		
Thermal Shrouds	180	77	77]	10.01.12.11.02
Gun Tube	2350	1960	1590		-70 LBS – DUAL AUTOFRETTAGE
Breech Actuator Assembly	N/A	43	43		-200 LBS - COMPOSITE -100 LBS - SUHSS
Breech Mech. Housing Assembly	(included)	42	42		
Breech Ring & Block Assemblies	1525	1,072	842	K	THREAD -50 LBS - TI BLOCK
_	(M1A2)	Estimate	Estimate		-180 LBS - FULL
	M256	SDD	LAEP		
				_	FS15149

M256 GUN WEIGHT DOES NOT INCLUDE ROTOR SHIELD LAEP WEIGHT SAVINGS OF 813 LBS

SDD ESTIMATE INCLUDES OTHER WEIGHT REDUCTION INITIATIVES NOT SHOWN IN PREVIOUS SLIDES

ARDEC

813 LBS

savings



LOS/BLOS ATD GUN ACCOMPLISHMENTS



ATD GUN MOD 1 FIRING DEVELOPMENTAL MRM CARTRIDGE 19 JUL 04 - SHOT 28





LOS/BLOS ATD GUN ACCOMPLISHMENTS - SUMMARY



ACCOMPLISHMENTS

- ACHIEVED A SYSTEM WEIGHT OF 4,240 POUNDS (ALL STEEL TUBE)
- IMPULSE WITHIN SPECIFIED LIMIT
- INITIATED TESTING DECEMBER 2003
- DEMONSTRATED A MUZZLE BRAKE EFFICIENCY OF 25%
- TEST FIRED TWO ITERATIONS OF BLAST DEFLECTOR TO REDUCE BLAST OVERPRESSURE
- TEST FIRED TWO ITERATIONS OF A COMPOSITE OVERWRAPPED GUN TUBE (JUL 04)
- SUCCESSFULLY PROOF TESTED AN ULTRA HIGH STRENGTH GUN STEEL TUBE (OCT 04)
- ACHIEVED TRL 6 IN OCTOBER 2004 (CANNON) WITH ALL STEEL
 TUBE



40Th Annual Armament Systems-Guns-Ammunition Rockets Missiles

Missile Systems Lethality Enhancement through the use of a Conducting Aerosol Plasma Warhead 27 April 2005

Allen Stults
US Army RDECOM AMRDEC
AMSRD-AMR-PS-WF
allen.stults@us.army.mil



Multi-Functional Warheads Are Lethal Against a Large Target Set

- Enhanced blast and fragmenting warheads have been successfully combined with shaped charges to service multiple target types with the same missile warhead, such as in Joint Common Missile. These are termed multi-purpose warheads.
- The next class of future missile systems can be further improved by adding RF effects to broaden the target set and enhance lethality
- The first step is to demonstrate additional effects without degrading existing capabilities



Multi-Effects Electromagnetic Warhead

Enhanced Blast for Personnel Lethality

Fragments for

Equipment Lethality

Develop and Integrate
Warheads into Missile
Systems to Destroy and
Disable Electronic Systems
and their Operators in
Support of Combat Forces

EMP for Electronics

Radio

Cell phone:

Compute

Vehicle Ignition System

GPS Jammer



Three Major Products From Missile Laboratory to Missile Programs

• Improved Anti-Armor Precursor Charge

 Enhanced GMLRS Bomblet or Cargo Round payload

• Next Generation 2.75" Rocket Warhead



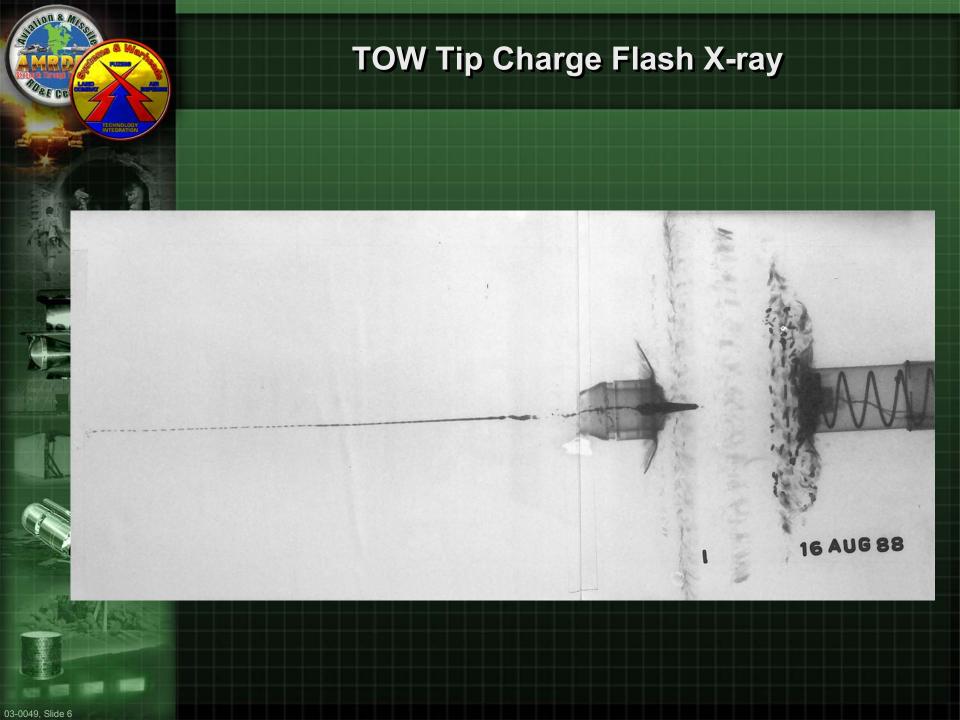
Baseline Warhead Trade Studies

 EM Measurements of Detonations and Plasma Characterizations

Non-Ideal Explosives for Shape Charges/
 Conductive Metal Antennas and Masonry destruction

Improved Seed Sources Program, SBIR leverage of FEG

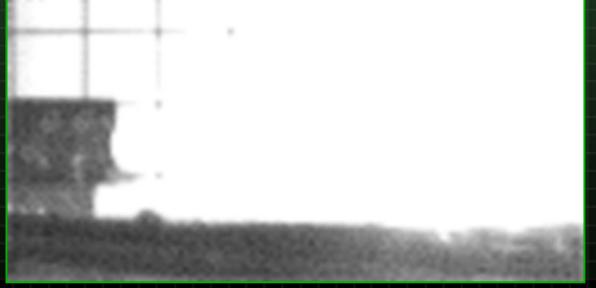
and FMG





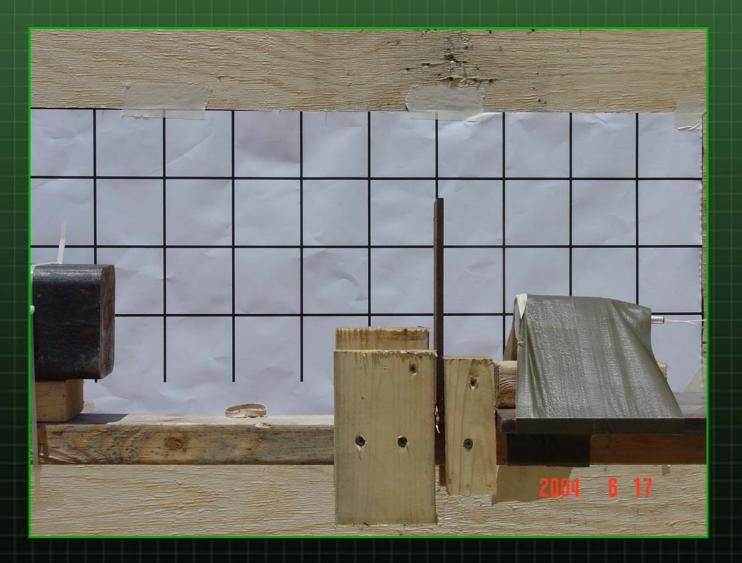
Plasma Entrainment







Test Set-Up





Test Article N5 and Metal Fuel Rich Mix





Filled test Article





Representative Cover Plates





Demonstration of Deflagration Effects on Masonry Targets

- Test various mixes of approximately 50 grams of aluminum powder pyrotechnic and thermite
- Gather fireball size, expansion rate and duration data for modeling efforts
- Investigating switch timing requirements to load dynamic plasma antenna
- Demonstrate robustness of masonry destructiveness at 10 CDs stand-off







Test 1 Video





Test 1 – High Speed Video



Test 1 – Masonry Destruction





























Conclusions and Plans

- Conductive Plasmas made from Deflagrating mixtures have significant destructive effects on masonry
- Significant Differences in Similar Mixes Allow the EM Designer a Robust set of Design Characteristics
- Temperature and Conductivity Effects will be further tested this Summer



Thanks

Joey Reed and Mike Kennemer for all their help in experimental conduct

George Arkoosh for his help in video production

Larry Altgilbers for his encouragement and advice



Royal Navy Small Calibre Gun Research to Defeat the Small Boat Threat

27th April 2005

Jonathan Watkins
Surface Warfare Weapons Team
Naval Systems
Dstl Portsdown West

[dstl]

3000 Staff
Based in a number of locations around the UK

Support to Capability
Management for Royal
Navy, British Army and
Royal Air Force.

UK Government Only Research and Technical Oversight of Research in Industry.







Dstl - Naval Systems Department

- The Naval Systems Department provides analysis and top-level platform and weapon systems advice in support of MOD decision making on Naval Systems.
- The Naval Systems Dept comprises of the following groups
 - Above Water Systems (Surface Warfare Weapon Systems Team)
 - Littoral Warfare (Operational Analysis)
 - ASW Capablility
 - Under Water Systems







Fast In Shore Attack Craft (FIAC)







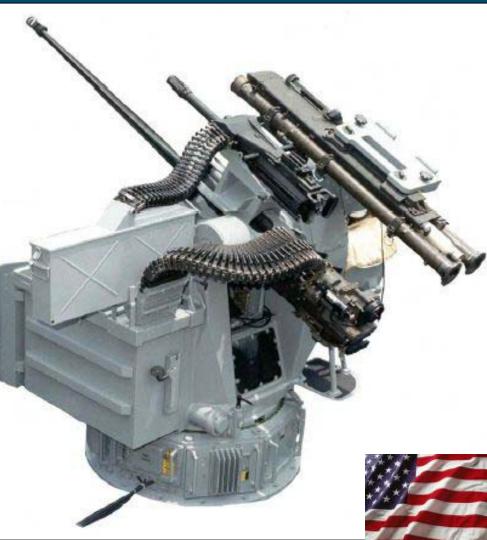






Existing Small Calibre Gun





e, used ements aft iary

(2km)

s by





Operator Performance?







HMS Somerset Trial

30mm Cannon - Remotely Controlled from Ops Room









Alternative Cannon



30x173mm MK44 Bush Master II



11 215

Proof Firings



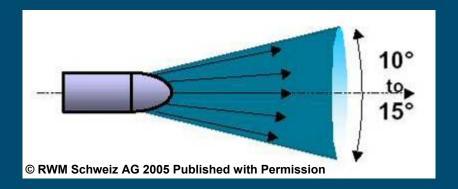


Air Burst Munitions

- Key Points for 30x170mm RWM Schweiz AG
 - 162 Sub-Projectile Kinetic Energy Payload
 - Each 1.24 g
 - Programmed to Eject Payload (Burst) Ahead of target







- Potential Advantages of ABM
 - Increased chance of hitting target due to better coverage by sub-projectile payload
 - Hence provides Increased lethality against soft targets





ABM Trial - Shoeburyness, Nov 2003

Co-operation with USN and USMC & Industry













QinetiQ





- Objectives of Trial
 - Assess ABM against representative target
 - Assess Penetration of fragments
 - Assess Fragment Dispersion
 - Assess Burst Point Placement





The Target Matrix

1km Target Matrix 11th November 2003







The Churchend Range

11th November 2003







ABM Burst Point Capture

1.5km Range





Front Camera

Side Camera



Target Plate Analysis

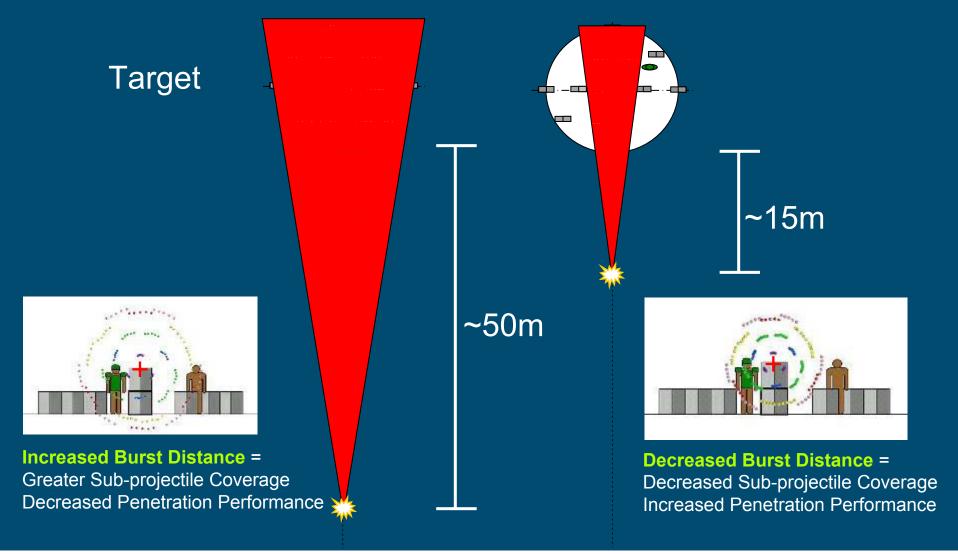








Effects of Burst Distance







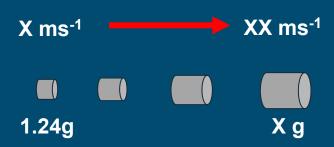
ABM Future Work

- Trial planned for May Jun 05
 - Different Design of ABM by General Dynamics (High Explosive Air Burst)
 - HEI rounds will be fired for direct comparison against target plates





 Gas Gun firings and Modelling to determine optimum sub-projectile size and associated lethality against threat set



- Results feed directly into both UK and US Navy Procurement Programmes
 - T23 Upgrade
 - US LPD17 & EFV





Potential Platforms for ABM



























Effects of High Explosive Rounds







Future Ammunition Work

- Investigate Lethality of a COTS range of Ammunition against precisely defined representative targets
- Larger calibres considered
- Using Typical threat materials and suitable position (e.g. angles)
- Determine required gun/ammunition lethality against the threat set









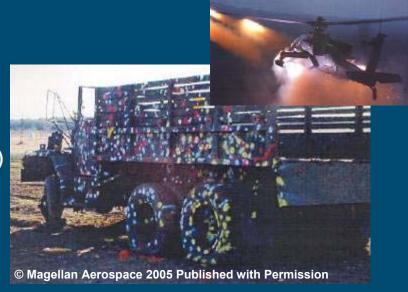


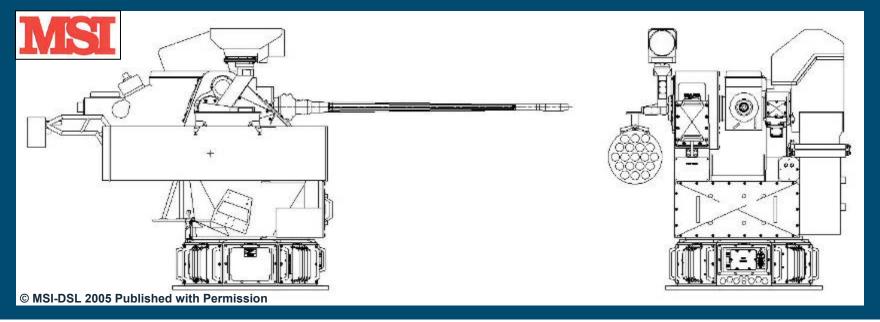




Hybrid Gun Mount

- 70mm Low Cost Rocket
- 6km Range (Increased with Guided Variant)
- Studies Conclude Launcher fit is feasible
 - Issues with Local Control Position

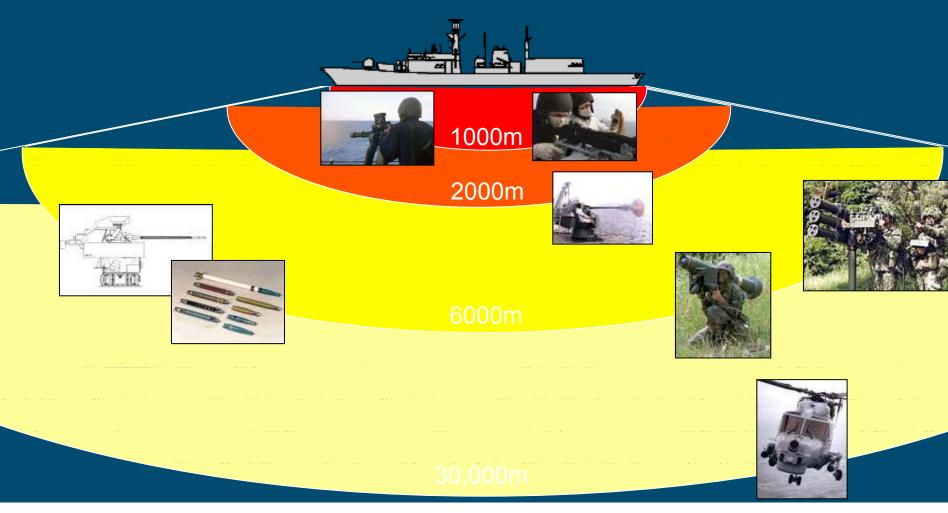








A Layered Defence







Implications of Swarm Attack

 Investigate Impact of dealing with a FIAC Swarm Attack from a SCGS

- Human Factors
 - Examine Human Computer
 Interface for Operator control
 - Prevent the operator from being overwhelmed?
- Can Technology assist?
 - Target Prioritisation?
 - BDA?



